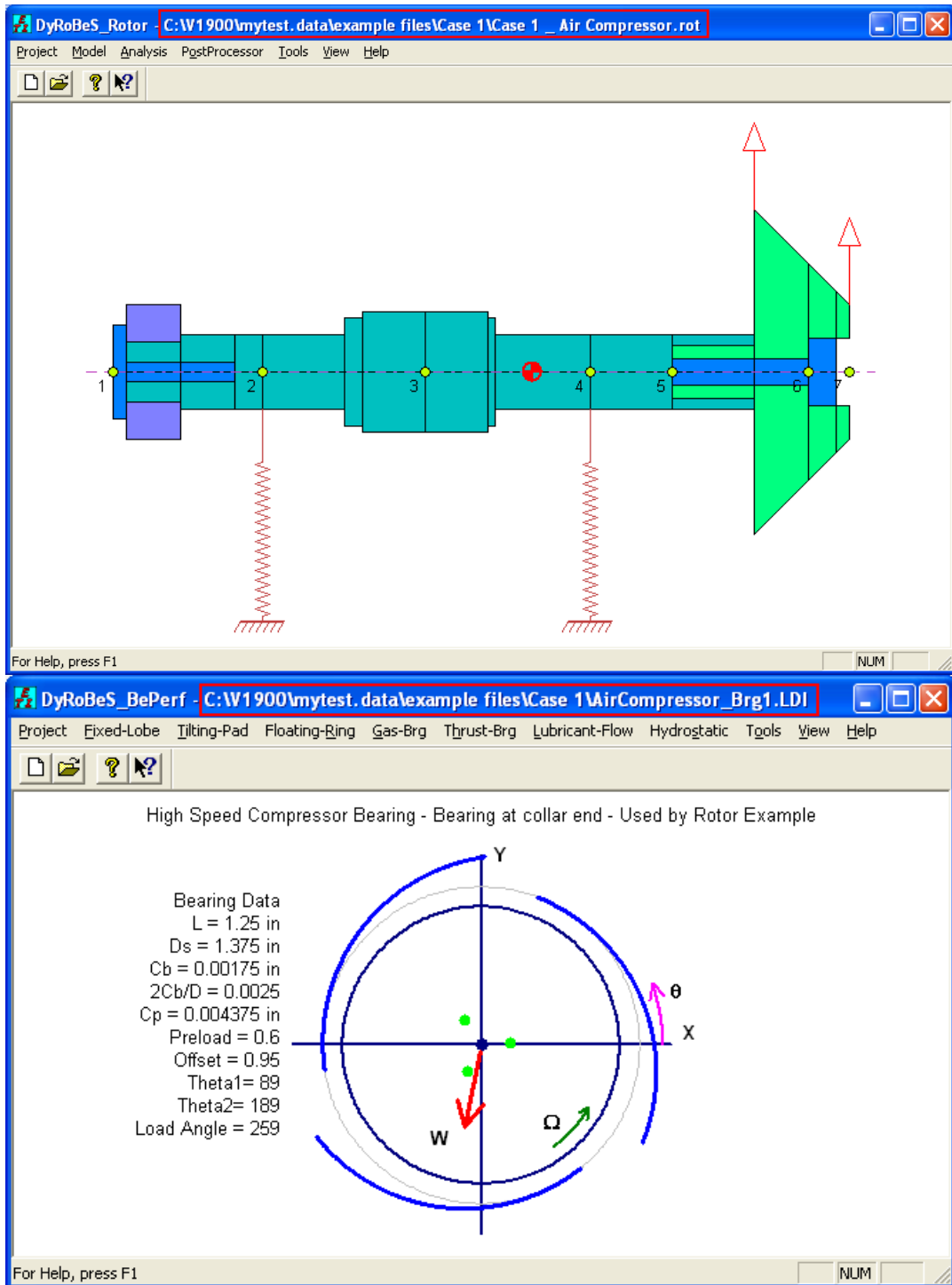


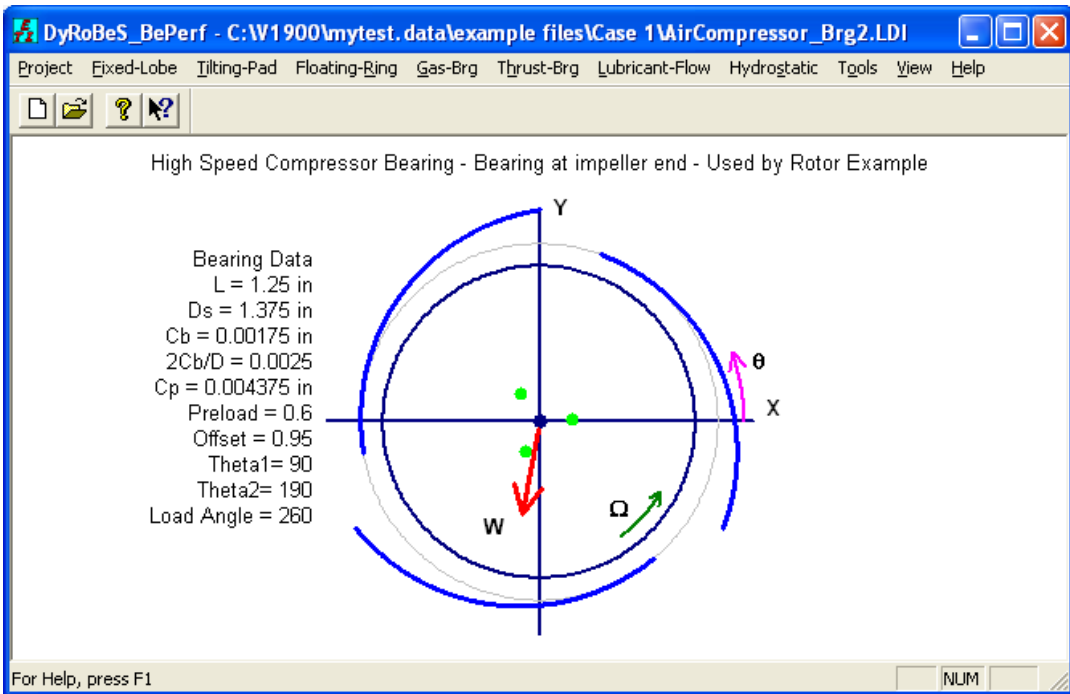
Examples for Ver 19.00

Three examples are used to demonstrate the new features implemented in Ver19.00 and some other commonly used features while performing the rotordynamics analysis.

Case 1 – Air Compressor

The rotor and bearing used in this example are shown below.





Fixed Pad Bearing - Dimensional Analysis

Comment: High Speed Compressor Bearing - Bearing at collar end - Used by Rotor Example

Coordinates: Standard Coordinates (X-Y) Load Angle: 259 degree

Bearing Type: 5 - Three Lobe K and C Coordinate Angle: 0 degree

Analysis Option: Heat Balance

Convert Units: English

Axial Length L: 1.25 (inch)

Journal Dia. D: 1.375 (inch)

Brg Radial Clr Cb: 0.00175 (inch)

Number of Pads: 3

Bearing Load = $W0 + W1 \times \text{RPM} + W2 \times \text{RPM}^2$ (Lbs)

W0: 145 W1: 0 W2: 0

Rotor Speeds (RPM) Additional Speeds

Start: 5000 End: 50000 Inc.: 5000

Lubricant: Amokon ISO-VG 32

Inlet Temperature: 120 (degF)

Heat carried away: 80 (%)

Bearing Data for Pad # 1

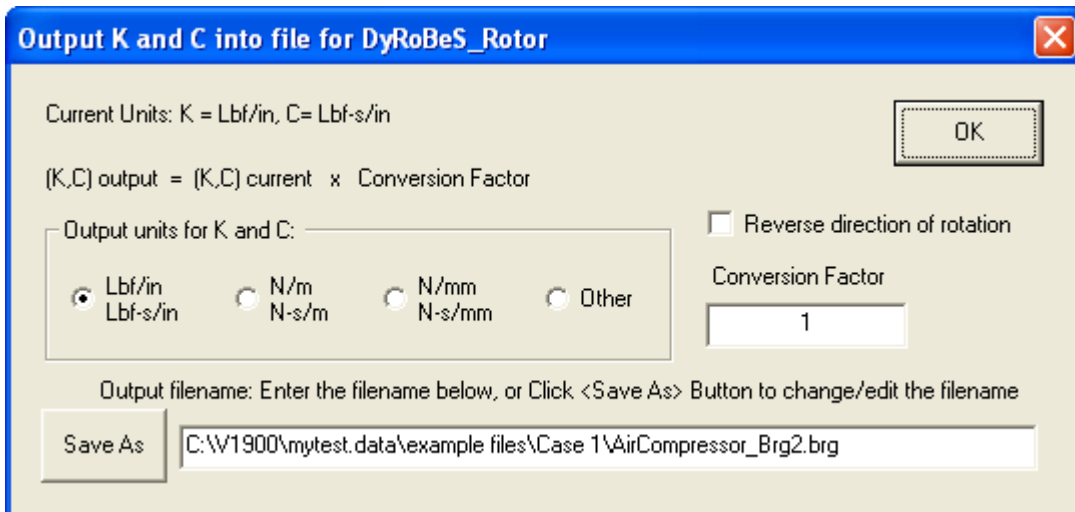
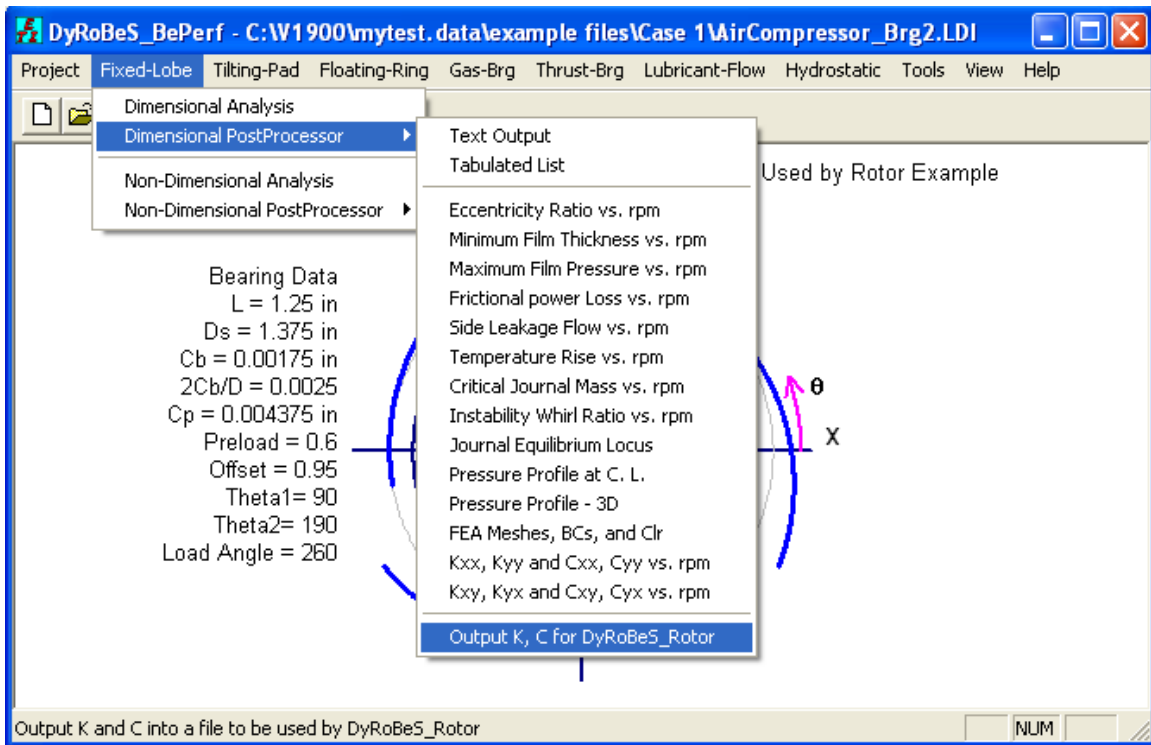
Leading Edge: 89 Preload: 0.6

Trailing Edge: 189 Offset: 0.95

Advanced Features
No

New Open Save Save As Run Close

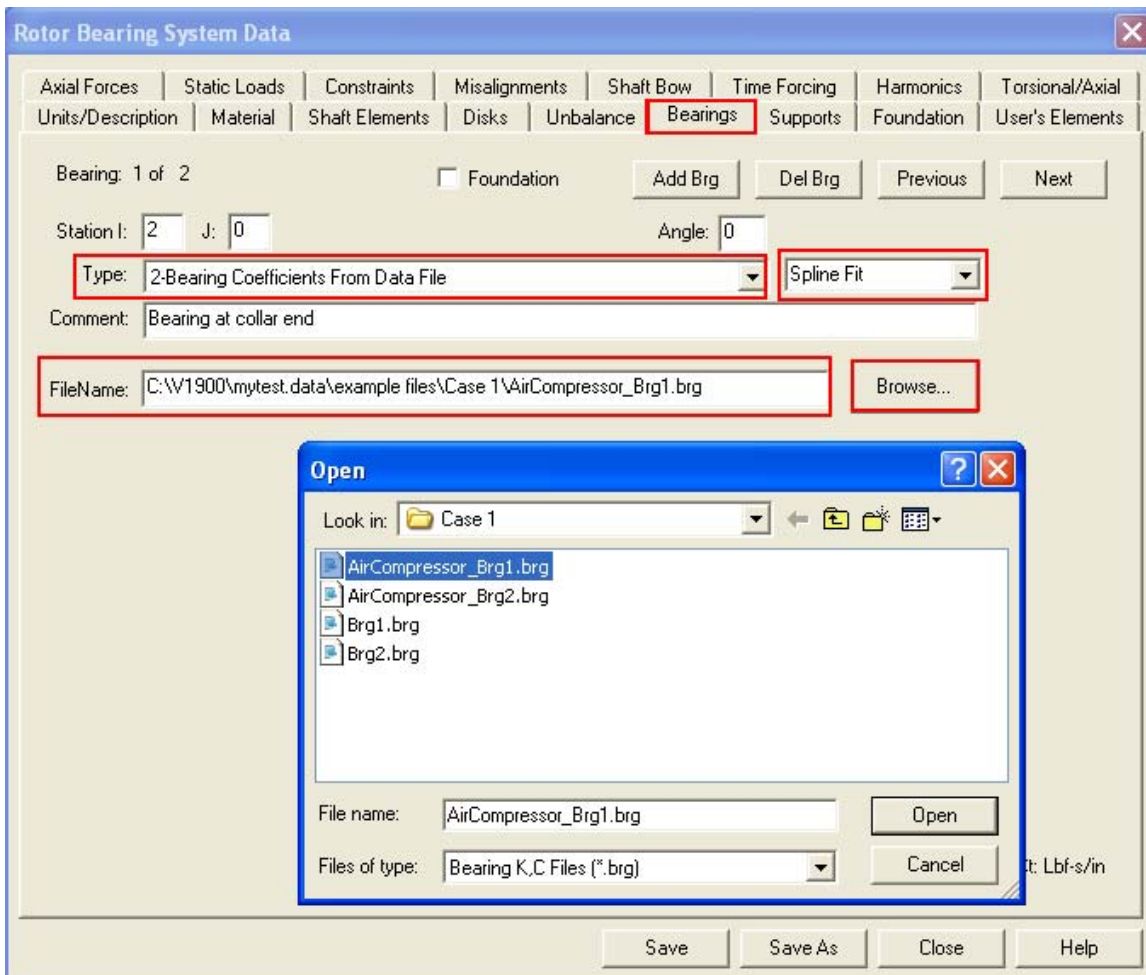
Note that the special characters, such as dot “.” and white space “ ”, are allowed in the file and path name. The bearing coefficients are calculated using BePerf from 5,000 to 50,000 rpm with an increment of 5,000 rpm. After the bearing run, the bearing coefficients can be outputted and then imported into the Rotor program.



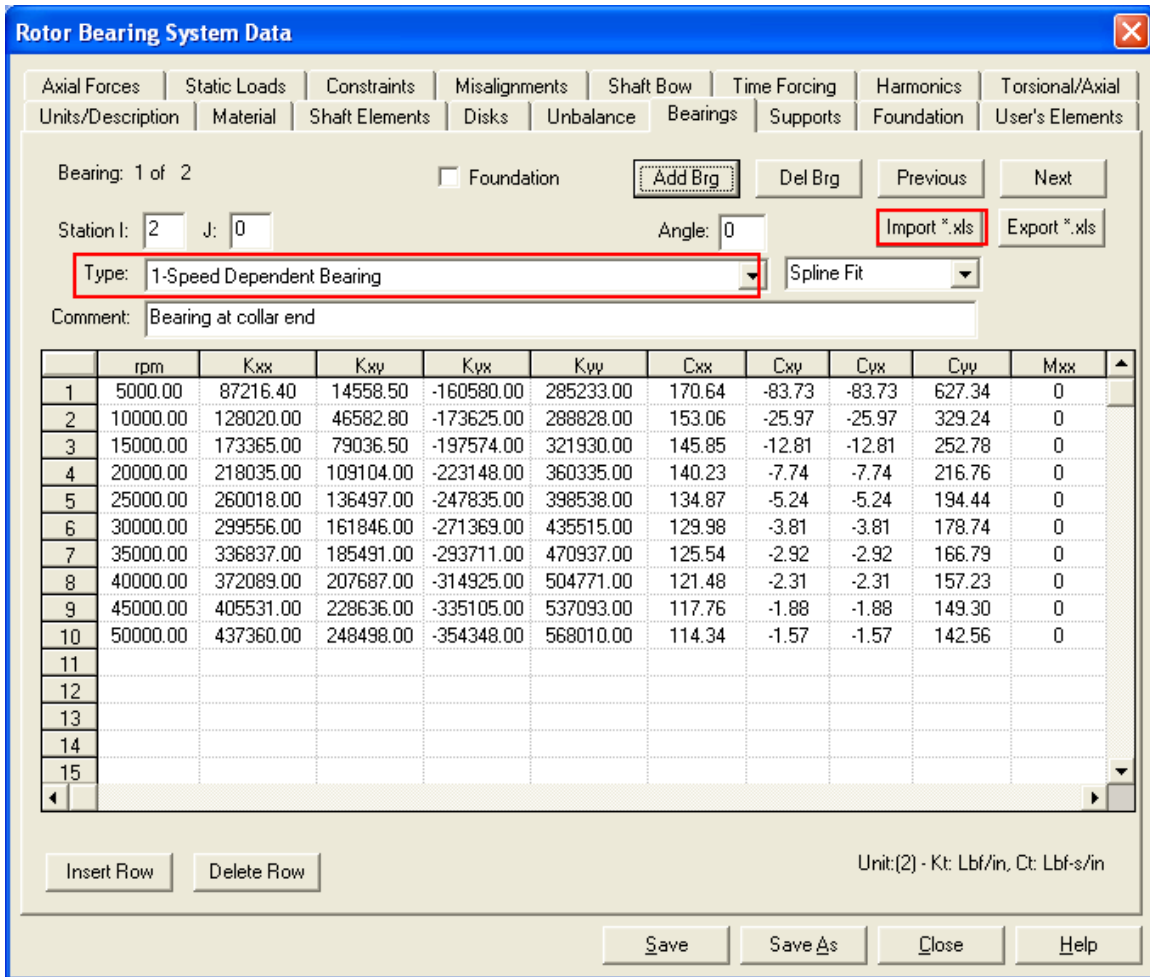
There is a Check Box named “Reverse direction of rotation” when outputting (saving) the bearing coefficients. Do **NOT** check this box unless you are performing a coupled Lateral-Torsional vibration analysis on a geared system and this particular shaft is rotating clockwise. Referring the previous bearing figure, all the bearing coefficients are calculated based on the shaft counter-clockwise rotation. For a single rotor system, this is always the case. For more details on the coupled lateral and torsional analysis, see book “Practical Rotordynamics and Fluid Film Bearing Design”, Chapter 8.

The bearing coefficients can be imported into the Rotor program using the Bearing Type 2 – Bearing Coefficients From Data File (Rotor – Model – Data Editor – Bearings –

Select Type 2 – then Browse the bearing file outputted from BePerf). In Ver 19.00, the interpolation method for the speed dependent bearing coefficients can be either spline function or linear interpolation. Spline function was used before Ver 19.00.



To keep the file simple and portable, we use Bearing Type 1- Speed Dependent Bearing and import the bearing coefficients from the BePerf output. If you have the bearing output file from BePerf program, this can be easily done by open the *.brg (output file from BePerf) using MS Excel, delete the first header line, save the file into xls format, then use the Import *.xls button in Rotor Bearing Tab to import the bearing coefficients. Note that this option is not commonly selected since you have all the necessary rotor and bearing data files and Bearing Type 2 is more suitable for this purpose. In some cases, if you obtain the bearing coefficients from other sources or if you need to share your rotor file but not the bearing data, then you can enter the bearing coefficients as the speed dependent bearing (Type 1).



The first analysis illustrated here is the Critical Speed Map, which is required by API. Note that the “spin/whirl ratio = 1” in the analysis input indicates the calculated critical speeds are the Synchronous Forward Critical Speeds commonly excited by the mass unbalance. The result can be viewed using the PostProcessor. The operating speed line and the bearing dynamic stiffness in the critical speed map can be plotted using the menu – Options – Settings. The bearing dynamic stiffness including damping effect is recommended in this map and required by the API. There are two intersection points (critical speeds) according to this map; one is around 23,000 rpm caused by K_{dx} and the other is around 26,000 rpm caused by K_{dy} . Due to significant damping in this case and the bearings are non-isotropic, there will only one critical speed around 26,000 rpm be observed in the unbalance response curve and machine test. In general, the higher one with more shaft flexibility will be observed.

Lateral Analysis Option & Run Time Data

Analysis: **3 - Critical Speed Map**

Shaft Element Effects
 Rotatory Inertia Shear Deformation Gyroscopic Gz

Static Deflection
 Constrained Bearing Stations

Critical Speed Analysis
 Spin/Whirl Ratio: 1
 No. of Modes: 5
 Brg Stiffness: **Kxxd dynamic**
 @ rpm: 23000

Critical Speed Map
 Spin/Whirl Ratio: 1
 Bearing K - Min: 1000
 Npts: 0 Max: 1e+008
 Stiffness to be varied at
 Bearings: All
 Allow Bearings in Series

Transient Analysis
 RPM: 35000 Time Domain Frequency Domain
 Constant Speed: 35000 rpm
 Time-Start: 0 Ending: 0.05 Increment: 1e-006
 Mass Unbalance Const. Unbalance Shaft Bow Disk Skew
 Gravity (X,Y) Gravity (Z)
 Static Loads Time Forcing Misalignment
 Solution Method: Newmark-beta Initial Cs: No

Gravity (g)
 X: 0 Y: -386.088 Z: 0
 None zero Gz Vertical Rotor

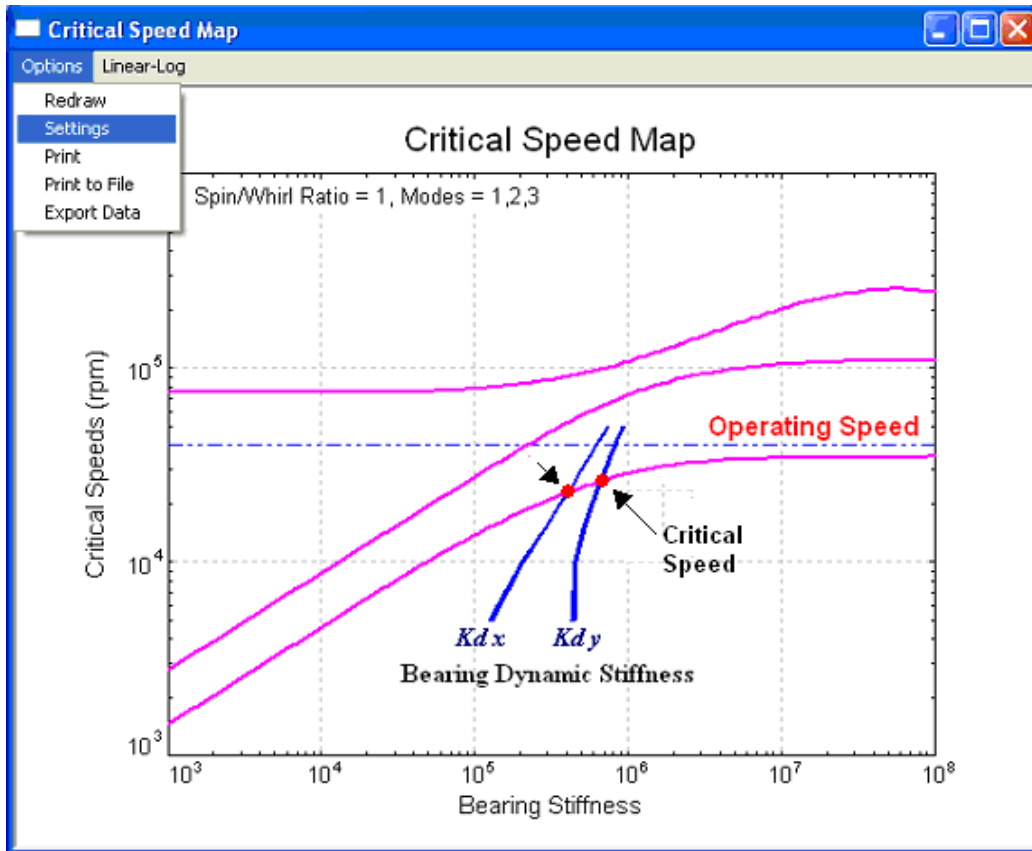
Whirl Speed and Stability Analysis
 RPM-Starting: 10000 Ending: 40000 Increment: 500 No. of Modes: 6

Steady State Synchronous Response Analysis
 RPM-Starting: 10000 Ending: 40000 Increment: 100 Excitation Shaft: 1
 All Synchronized Shafts
 Effects:
 Mass Unbalance Const. Unbalance Shaft Bow Disk Skew Misalignment

Steady State Harmonic Excitation
 RPM-Starting: 10000 Ending: 40000 Increment: 100 Excitation Shaft: 1
 All Shafts with same speed

Steady Maneuvers (Base Constant Translational Acceleration and/or Turn Rate)
 Speed (RPM): 35000 Acceleration - X: 0 Y: 772.176 Turn Rate - X: 0 Y: 0 Ref Pos: 0

Run Cancel



Critical Speed Map

Title:

X-Label:

Y-Label:

Modes: For example: 1,2,3,5,7,9 or 3

For Selected Modes, enter mode numbers separated by commas.

Operating Speed Range: to

Print Spin/Whirl Ratio

Curve Fit Symbol Major Grid Minor Grid

Manual Scaling

Manual Scaling Data

Xmin: Ymin:

Xmax: Ymax:

XDiv: YDiv:

Bearing Stiffness

Import from the existing DyRoBeS Brg files

Direct Stiffness - K Dynamic Stiffness - $K^2 + (Cw)^2 \wedge 0.5$

Color Number:

	rpm-1	Kxx-1	Kyy-1	rpm-2	Kxx-2	Kyy-2	rpm-3	Kxx-3	Kyy-3
1	5000	124859	435031	5000	130387	456826	0	0	0
2	10000	205136	449768	10000	207937	463477	0	0	0
3	15000	287306	511170	15000	288731	521913	0	0	0
4	20000	365775	579603	20000	366399	588808	0	0	0
5	25000	438488	646492	25000	438590	654763	0	0	0
6	30000	506447	710620	30000	506174	718254	0	0	0
7	35000	570240	771689	35000	569684	778857	0	0	0
8	40000	630380	829787	40000	629601	836605	0	0	0
9	45000	687313	885132	45000	686355	891667	0	0	0
10	50000	741421	937962	50000	740316	944269	0	0	0
11									
12									
13									
14									
15									

To examine the critical speed mode shapes, the critical speed analysis is performed. First, let us examine the critical speed around 26,000 rpm estimated from the Critical Speed Map. In Ver 19.00, the **dynamic** stiffness option is added and the speed can be specified. Before Ver 19, only the **direct** stiffness and the bearing coefficients from the Last speed point (for the speed dependent coefficients) were used. For compatible with the old versions, the speed ZERO can be entered to indicate the last speed point will be used. From the energy distribution, the bearing at impeller end has much more potential energy than that of the collar end (33.7% vs. 5.2%). Therefore, this bearing is more critical in the design. For the critical speed around 23,000 rpm (in the x-direction), enter 23.000 rpm and select *Kxxdynamic* in the stiffness option. Most of the potential energy in this case is in the bearings and the shaft only has 36% potential energy. This mode is more close to the rigid bode mode compared with the mode at the 26,000 rpm, therefore, this mode is not observed in the unbalance response curve, which will be shown later.

In Ver 19, the bearing coefficients after interpolation and used in the analysis are also printed in the text output for reference. A warning message will appear if the analysis speed is outside the speed range of the bearing coefficients. Again, in the Critical Speed Analysis, the ZERO speed indicates the last speed point will be used as before ver 19.

```

*****
Brg Coefficients after Interpolation
No. rpm      Kxx      Kxy      Kyx      Kyy      Cxx      Cxy      Cyx      Cyy
1 26000.    268110.   141715.  -252637.  406053.   133.853  -4.91026  -4.91026  190.982
2 26000.    269498.   137247.  -259252.  410936.   133.480  -6.44488  -6.44488  193.375
=====
***** Critical Speed Analysis *****

    *** Y-Direction Properties are specified ***
    Dynamic Stiffness = sqrt(Kyy^2+(omegaCyy)^2)

    Bearing Coefficients Used @ Speed = 26000.00 RPM

    Bearing No: 1      Stiffness Kt = 659747.46      Kr = 0.0000000
    Bearing No: 2      Stiffness Kt = 667890.00      Kr = 0.0000000

    ***** Spin(1)/Whirl Ratio = 1.000 *****

    no      rpm      R/S      Hz
    1      26408.9    2765.53    440.148
    2      63294.6    6628.20    1054.91
    3      98313.0    10295.3    1638.55
    4      302000.    31625.3    5033.33
    5      590132.    61798.5    9835.53
    6      0.101755E+07  106558.    16959.2
    7      0.176134E+07  184447.    29355.7
    8      0.342898E+07  359082.    57149.7
    9      0.463345E+07  485214.    77224.2

    *****

```


Lateral Analysis Option & Run Time Data

Analysis: 2 - Critical Speed Analysis

Shaft Element Effects
 Rotatory Inertia Shear Deformation Gyroscopic Gz

Static Deflection
 Constrained Bearing Stations

Critical Speed Analysis
 Spin/Whirl Ratio: 1
 No. of Modes: 5
 Brg Stiffness: **Kyyd dynamic** @ rpm: 26000

Critical Speed Map
 Spin/Whirl Ratio: 1
 Bearing K - Min: 1000
 Npts: 0 Max: 1e+008
 Stiffness to be varied at

Transient Analysis
 RPM: 35000 Time Domain Frequency Domain
 Constant Speed: 35000 rpm
 Time-Start: 0 Ending: 0.05 Increment: 1e-006
 Mass Unbalance Const. Unbalance Shaft Bow Disk Skew Gravity (X,Y) Gravity (Z)
 Static Loads Time Forcing Misalignment

Gravity (g)
 X: 0 Y: -386.088 Z: 0
 None zero Gz Vertical Rotor

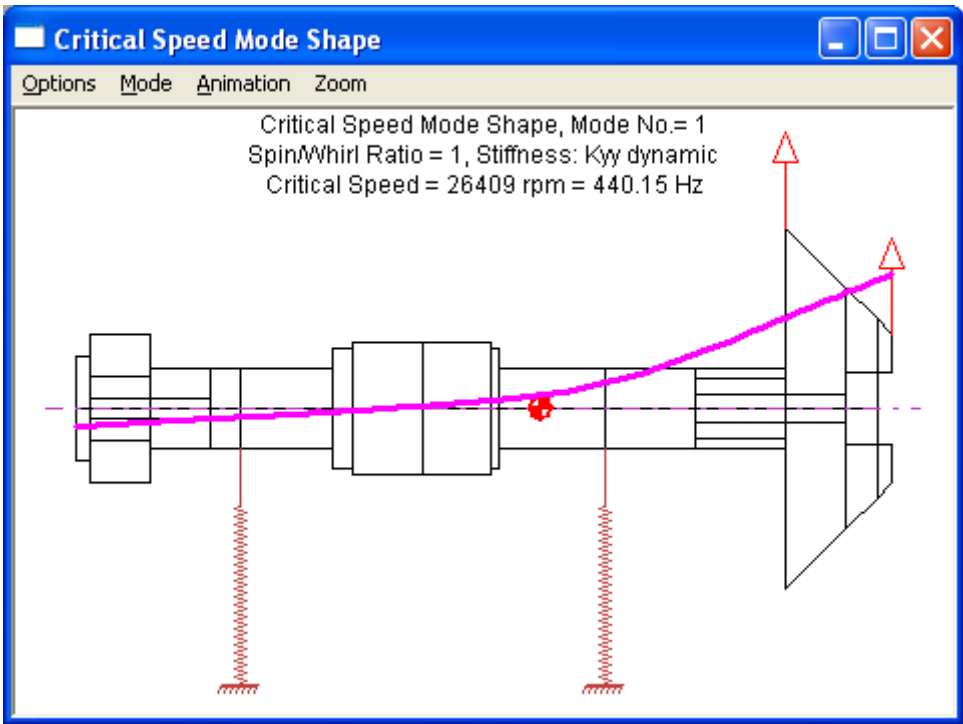
Whirl Speed and Stability Analysis
 RPM-Starting: 10000 Ending: 40000 Increment: 500 No. of Modes: 6

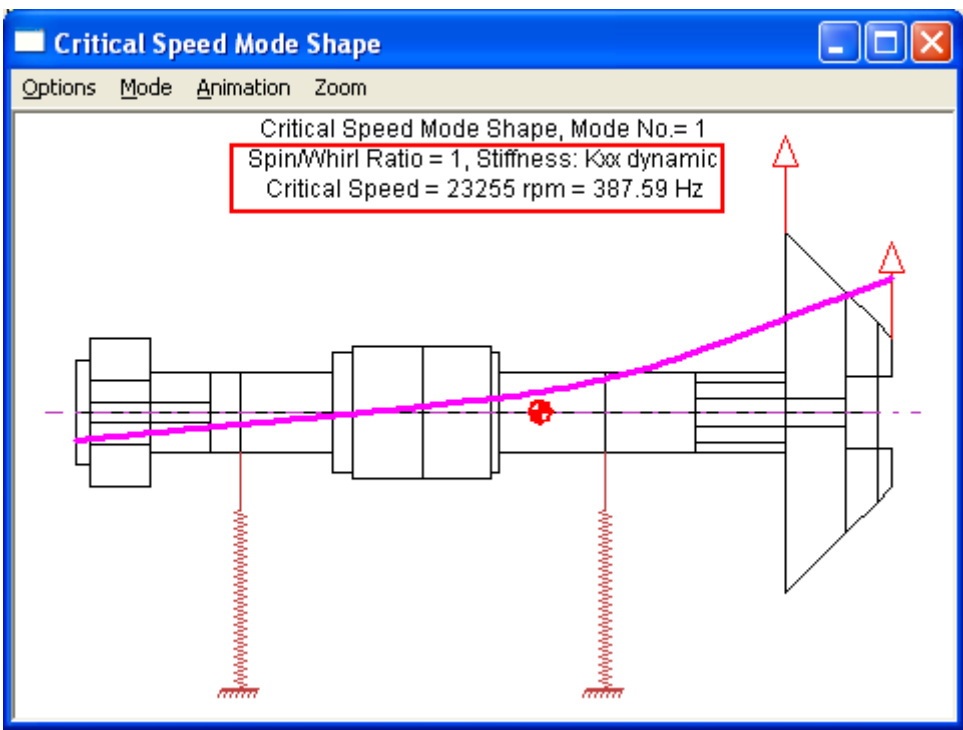
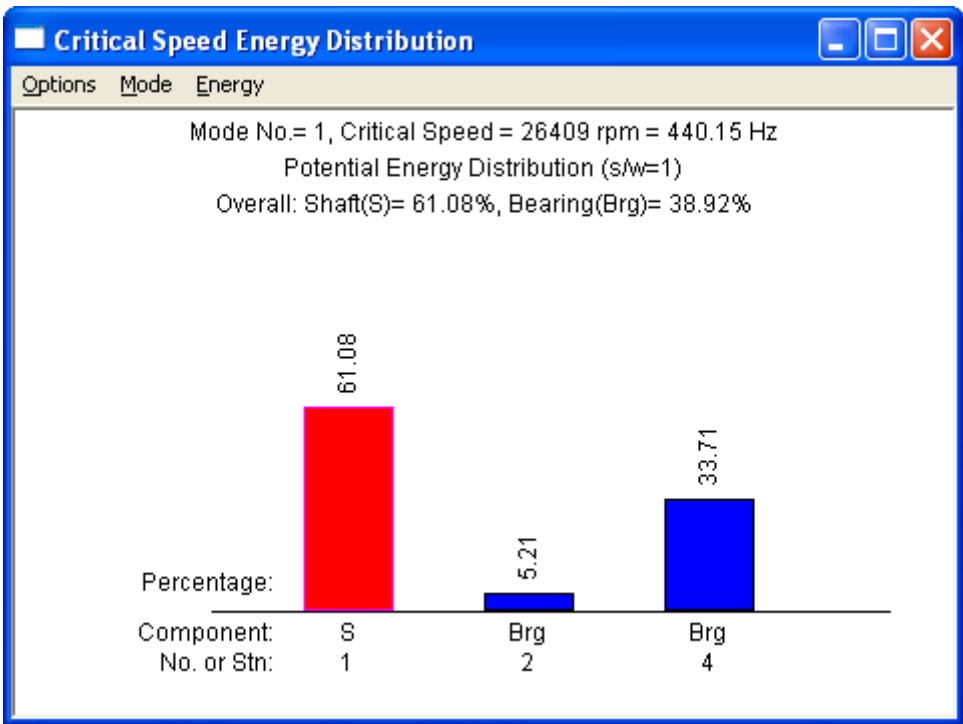
Synchronous Response Analysis
 Ending: 10000 Increment: 100 Excitation Shaft: 1
 All Synchronized Shafts

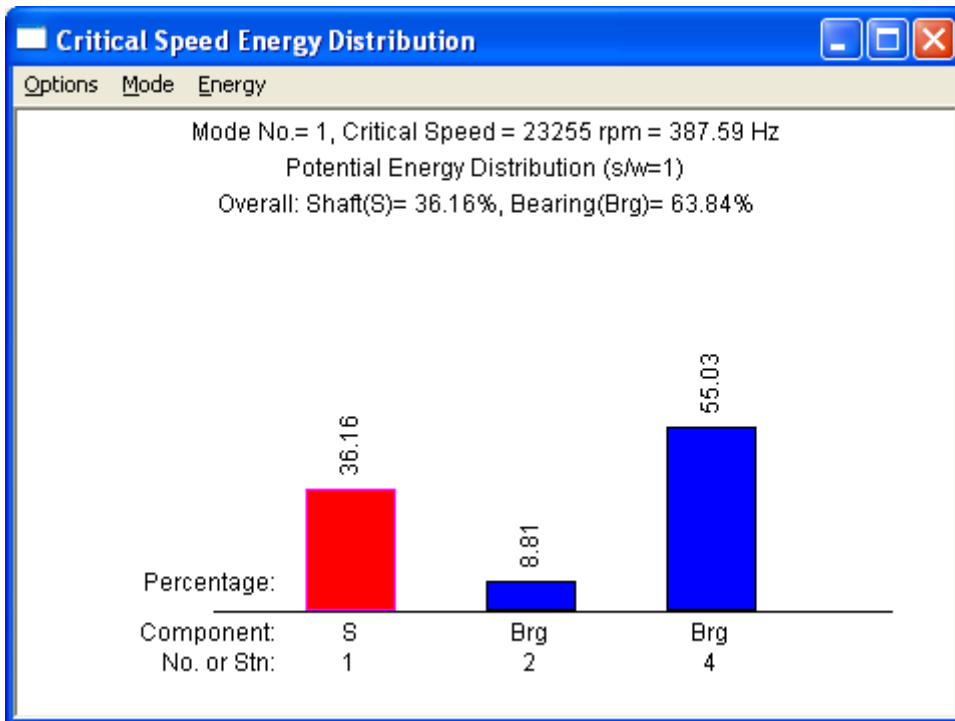
Steady State Harmonic Excitation
 RPM-Starting: 10000 Ending: 40000 Increment: 100 Excitation Shaft: 1
 All Shafts with same speed

Steady Maneuvers (Base Constant Translational Acceleration and/or Turn Rate)
 Speed (RPM): 35000 Acceleration - X: 0 Y: 772.176 Turn Rate - X: 0 Y: 0 Ref Pos: 0

Buttons: Run, Cancel







The next analysis is the unbalance response analysis. The analysis speed starts from 10,000 rpm to 40,000 rpm with an increment of 100 rpm. Since this speed increment (100 rpm) is smaller than the speed increment (5,000 rpm) in the bearing coefficients, interpolation will be implemented in the program. The interpolation method (spline or linear) is specified in the Bearing input tab. The vibration at the probe station (station 5) is shown in the Bode plot (PostProcessor – Steady Synchronous Response – Bode Plot). Some graphic settings can be adjusted under Options – Settings. In this example, English units are used. The response displacement is in inches. But, it is common to use mils in US to describe the vibration amplitude. Therefore, we entered 1000 in the Amplitude Scale to convert the inches to mils. The Text Color box is checked to show the corresponding color in the text printout in the plot. The curve color can also be changed in the Preference Settings – Post-Processor Graph Colors. Also, the amplitude printout is formatted with floating point with 2 decimal points. Some of the graphic settings can be pre-defined and saved in the Preference Settings File (Project – Preference Settings – Post-Processor Graph Settings and Post-Processor Graph Colors). These graphic features are implemented in all the post-processor plots when applicable. Again, use Options – Settings to make necessary adjustments to meet your needs.

Lateral Analysis Option & Run Time Data

Analysis: **5 - Steady State Synchronous Response - Linear System**

Shaft Element Effects
 Rotatory Inertia Shear Deformation Gyroscopic Gz

Static Deflection
 Constrained Bearing Stations

Critical Speed Analysis
 Spin/Whirl Ratio: 1
 No. of Modes: 5
 Brg Stiffness: Kyzd dynamic
 @ rpm: 26000

Critical Speed Map
 Spin/Whirl Ratio: 1
 Bearing K - Min: 1000
 Npts: 0 Max: 1e+008
 Stiffness to be varied at
 Bearings: All
 Allow Bearings in Series

Transient Analysis
 RPM: 35000 Time Domain Frequency Domain
 Constant Speed: 35000 rpm
 Time-Start: 0 Ending: 0.05 Increment: 1e-006
 Mass Unbalance
 Const. Unbalance
 Shaft Bow
 Disk Skew
 Gravity (X,Y)
 Gravity (Z)
 Static Loads
 Time Forcing
 Misalignment
 Solution Method: Newmark-beta
 Initial Cs: No

Gravity (g)
 X: 0
 Y: -386.088
 Z: 0
 None zero Gz
 Vertical Rotor

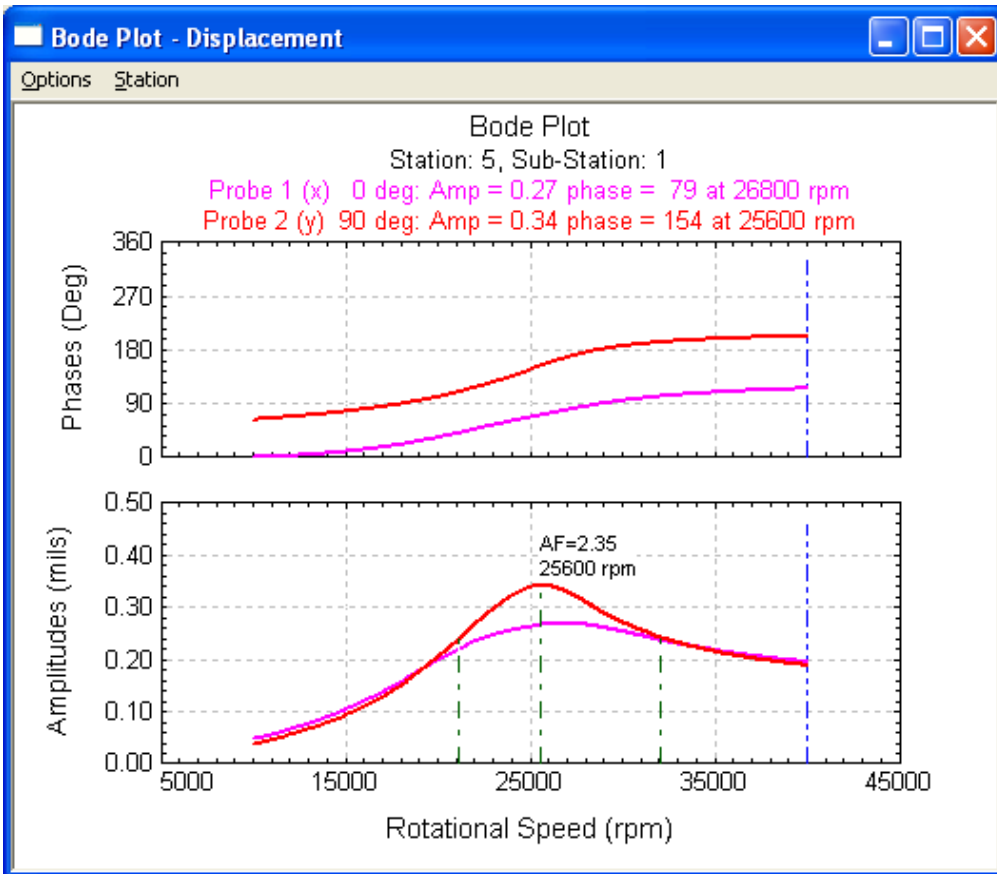
Whirl Speed and Stability Analysis
 RPM-Starting: 10000 Ending: 40000 Increment: 500 No. of Modes: 6

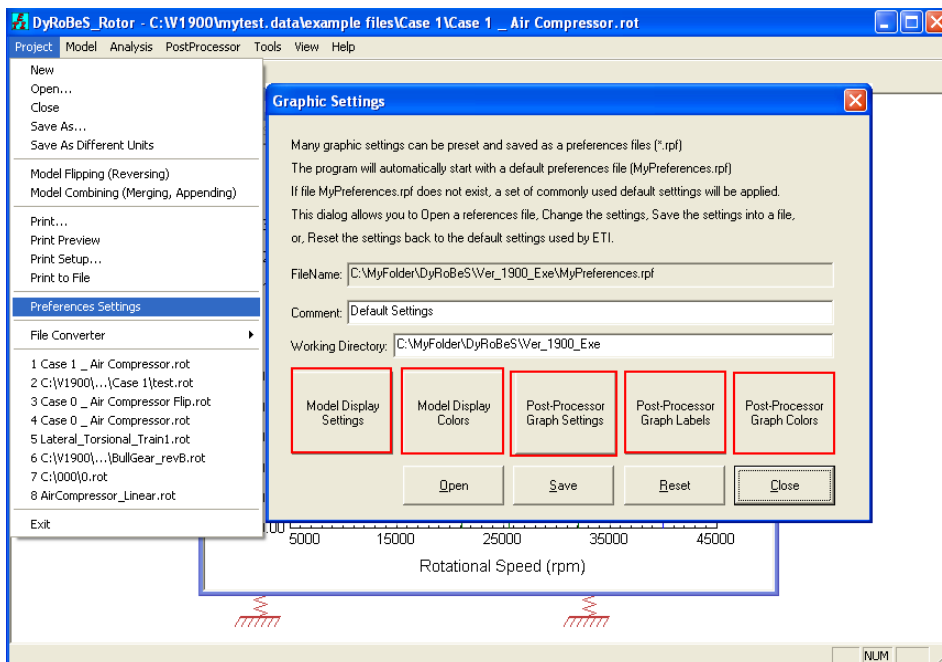
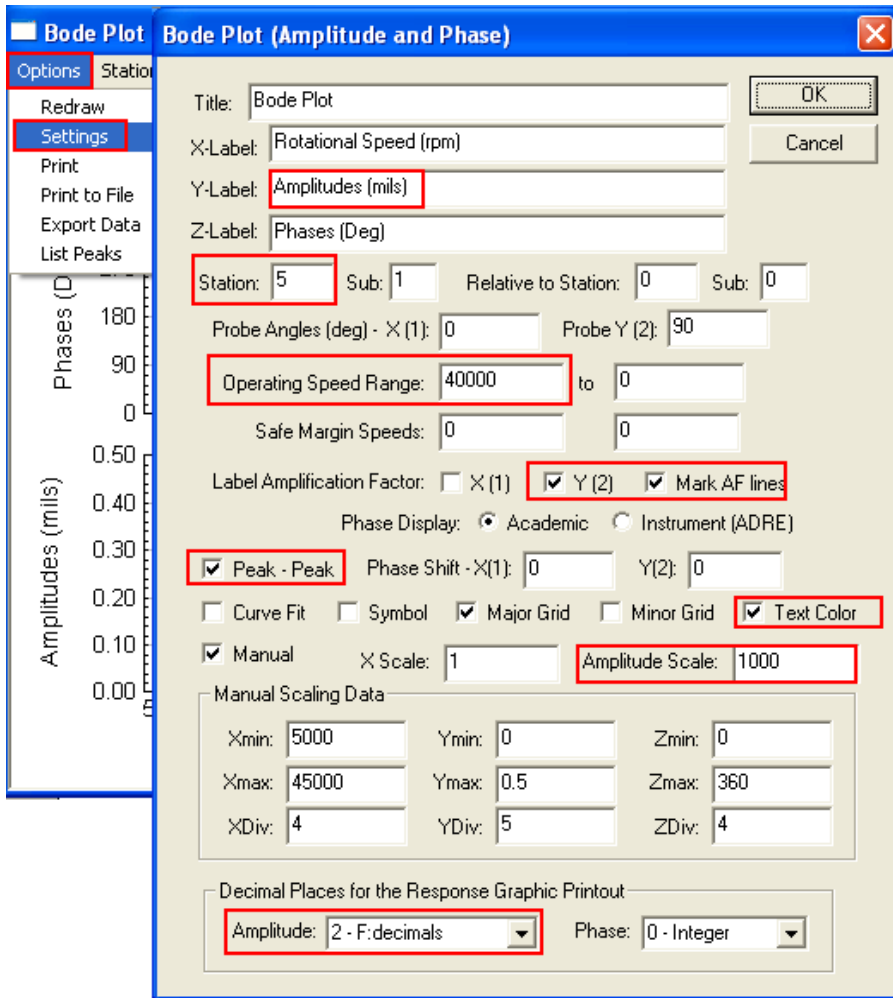
Steady State Synchronous Response Analysis
 RPM-Starting: 10000 Ending: 40000 Increment: 100
 Excitation Shaft: 1
 All Synchronized Shafts
 Effects:
 Mass Unbalance
 Const. Unbalance
 Shaft Bow
 Disk Skew
 Misalignment

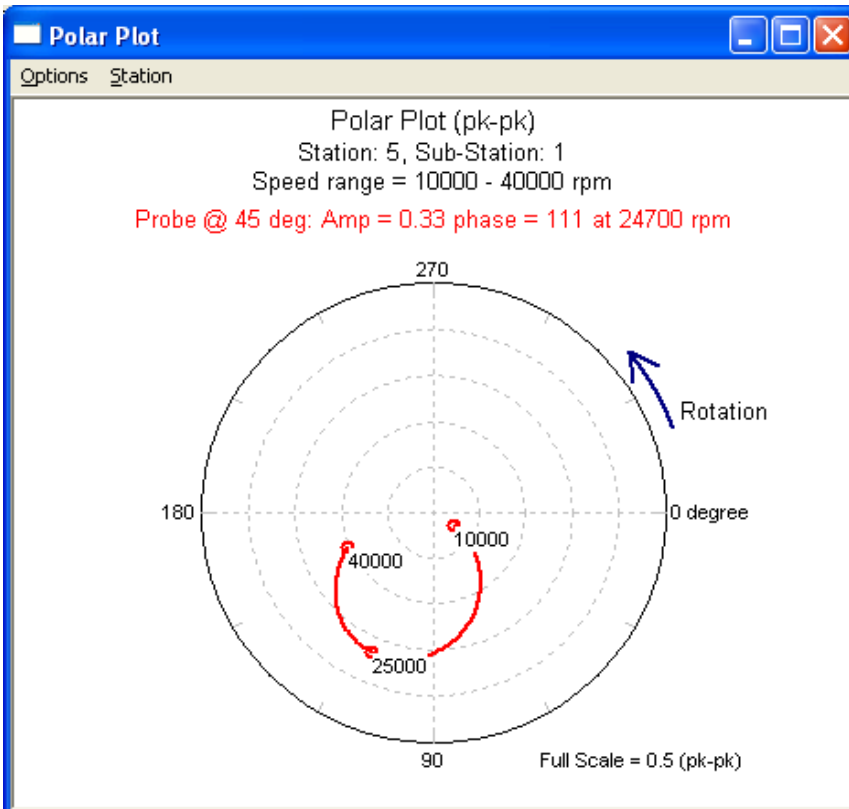
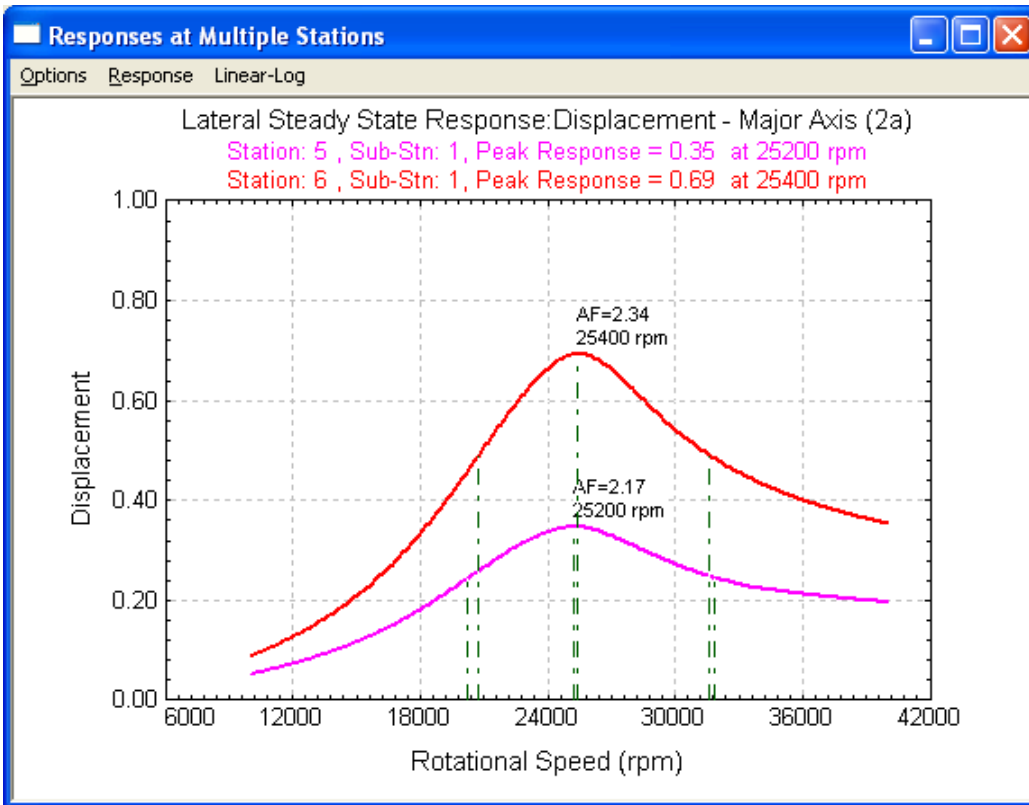
Steady State Harmonic Excitation
 RPM-Starting: 10000 Ending: 40000 Increment: 100
 Excitation Shaft: 1
 All Shafts with same speed

Steady Maneuvers (Base Constant Translational Acceleration and/or Turn Rate)
 Speed (RPM): 35000 Acceleration - X: 0 Y: 772.176 Turn Rate - X: 0 Y: 0 Ref Pos: 0

Run Cancel

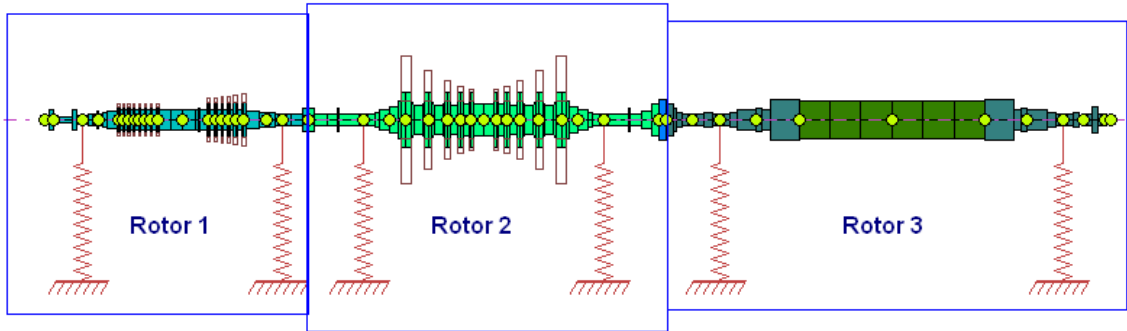




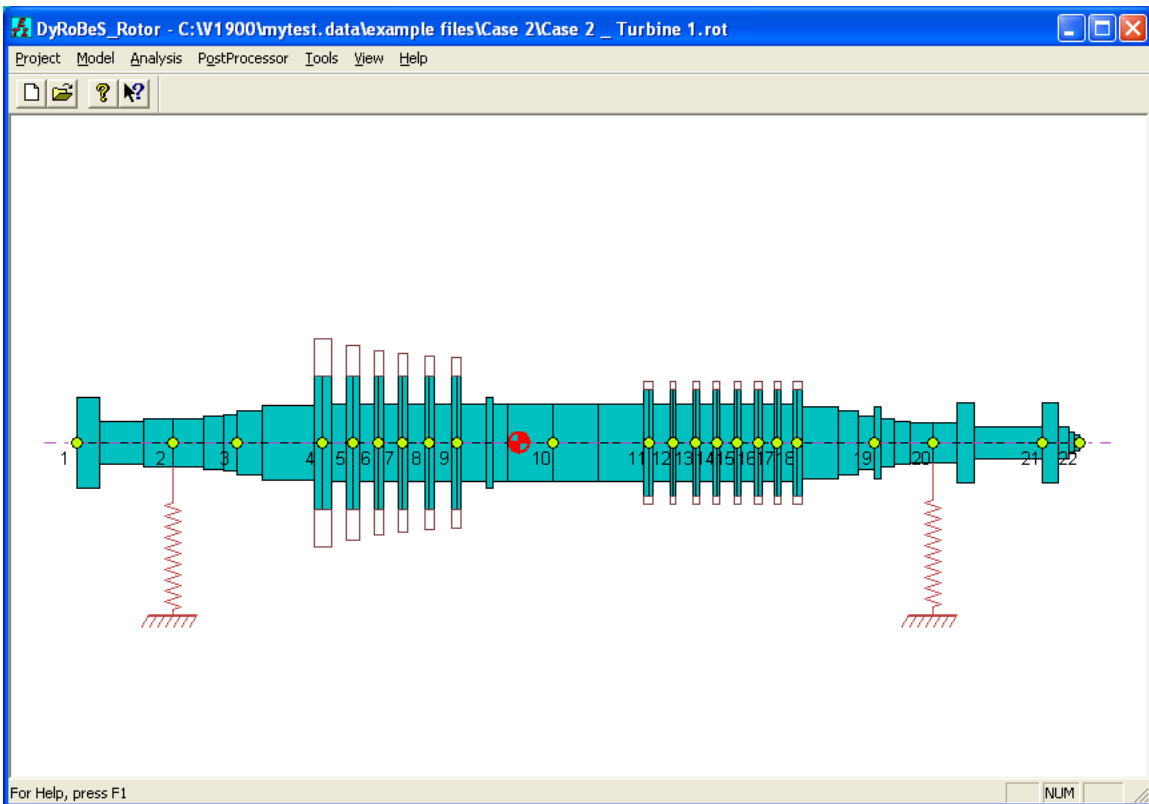


Case 2 – Turbine – Generator Set

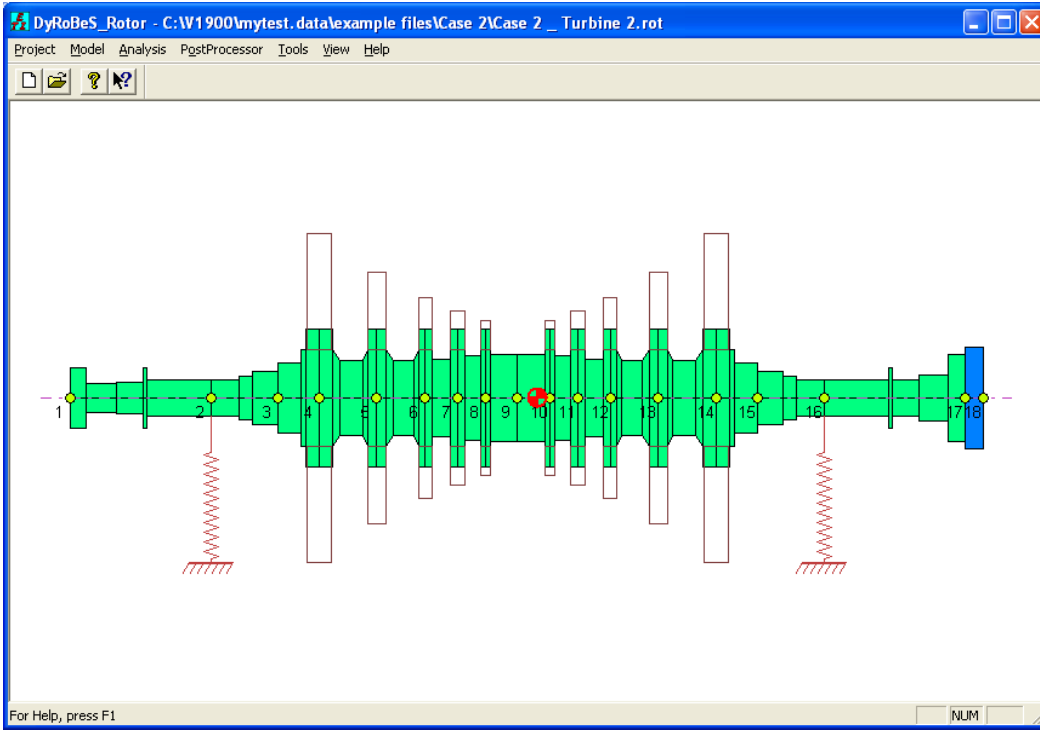
This example is a turbine-generator set. The model was built in three separate rotors (files) as shown below. It is noticed that the individual Rotor 1 was modeled in the reversed direction (right to left) compared to the entire train. This can occur frequently since all the drawings (data or model) were provided from different suppliers, or modeled by different engineers. In this example, the Model Flipping and Model Combining features will be demonstrated.



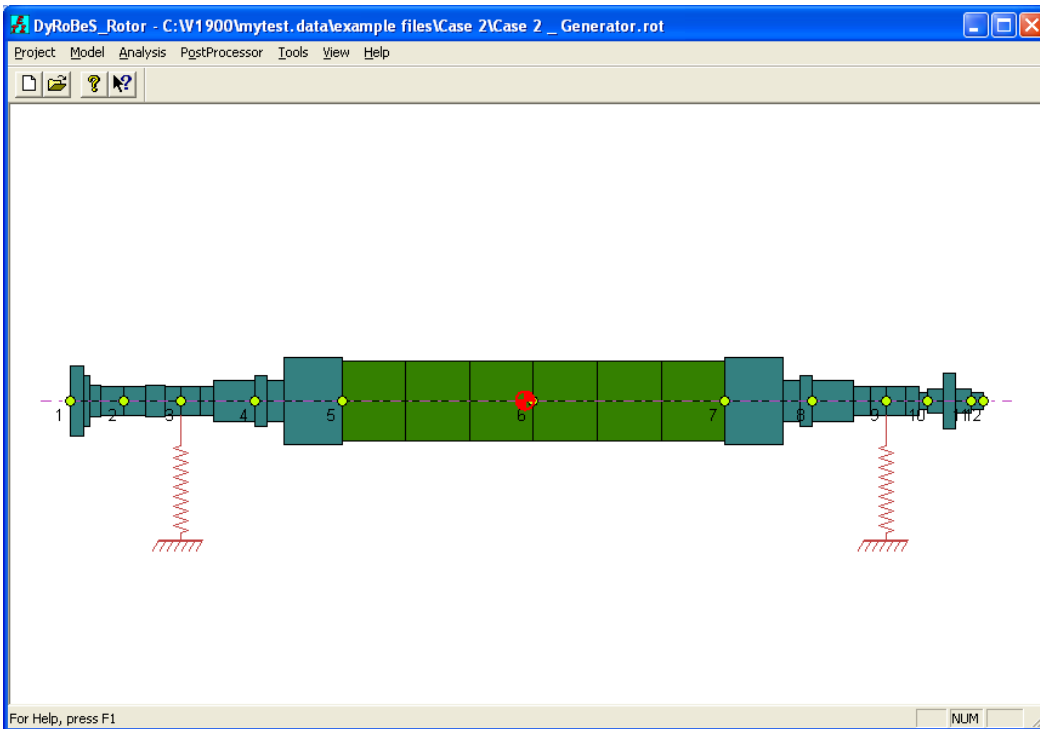
Rotor 1



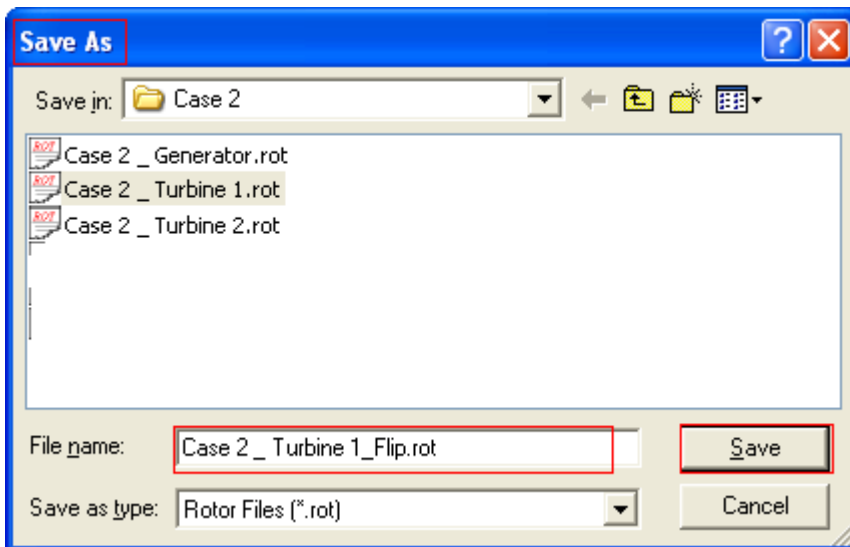
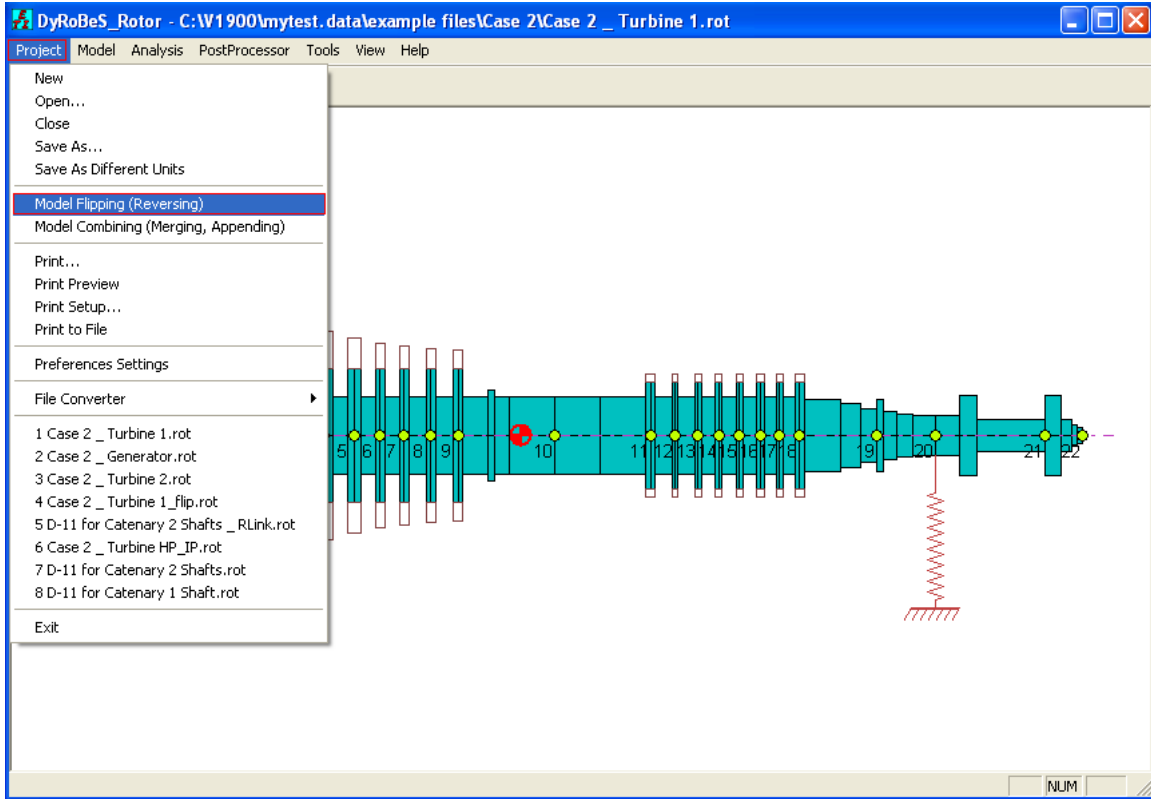
Rotor 2

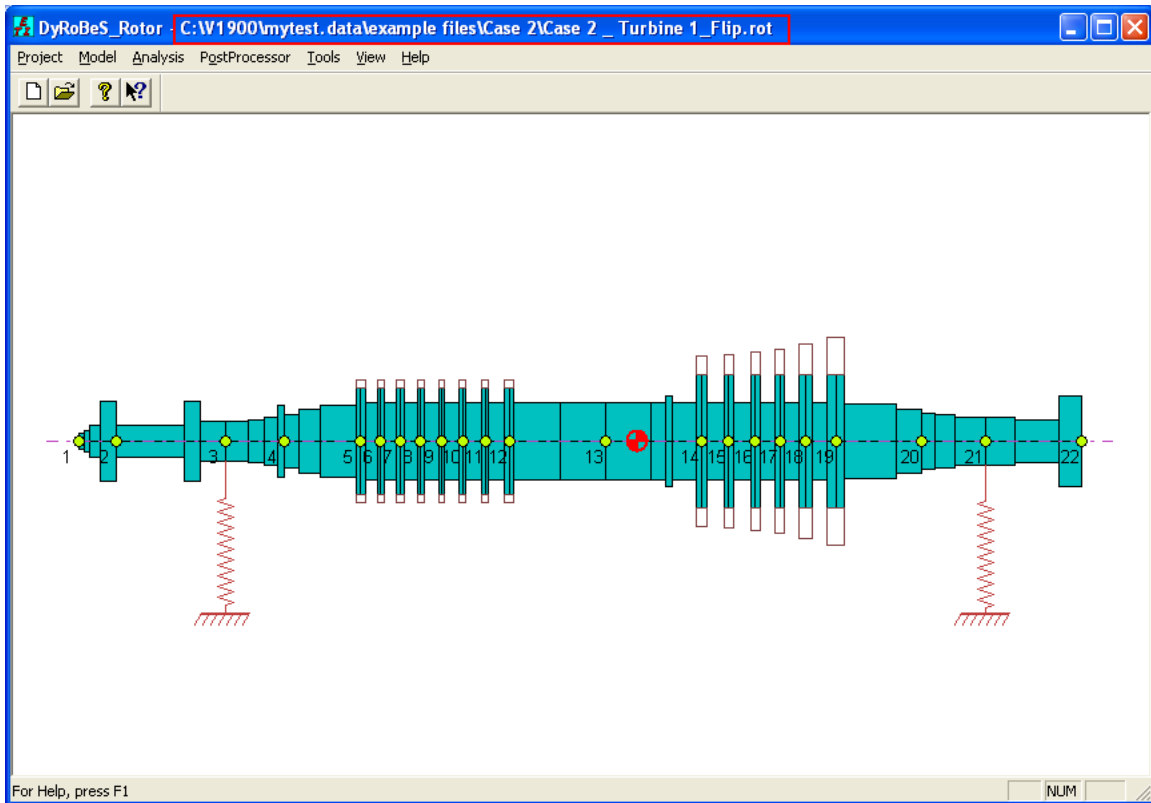


Rotor 3

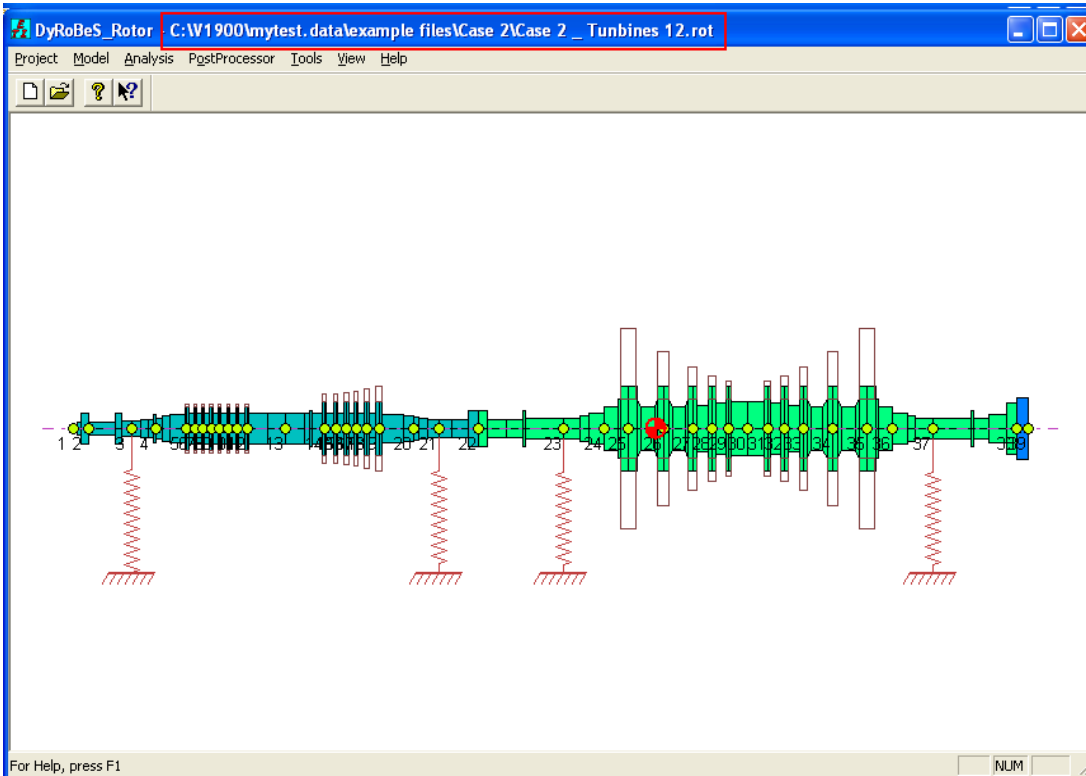
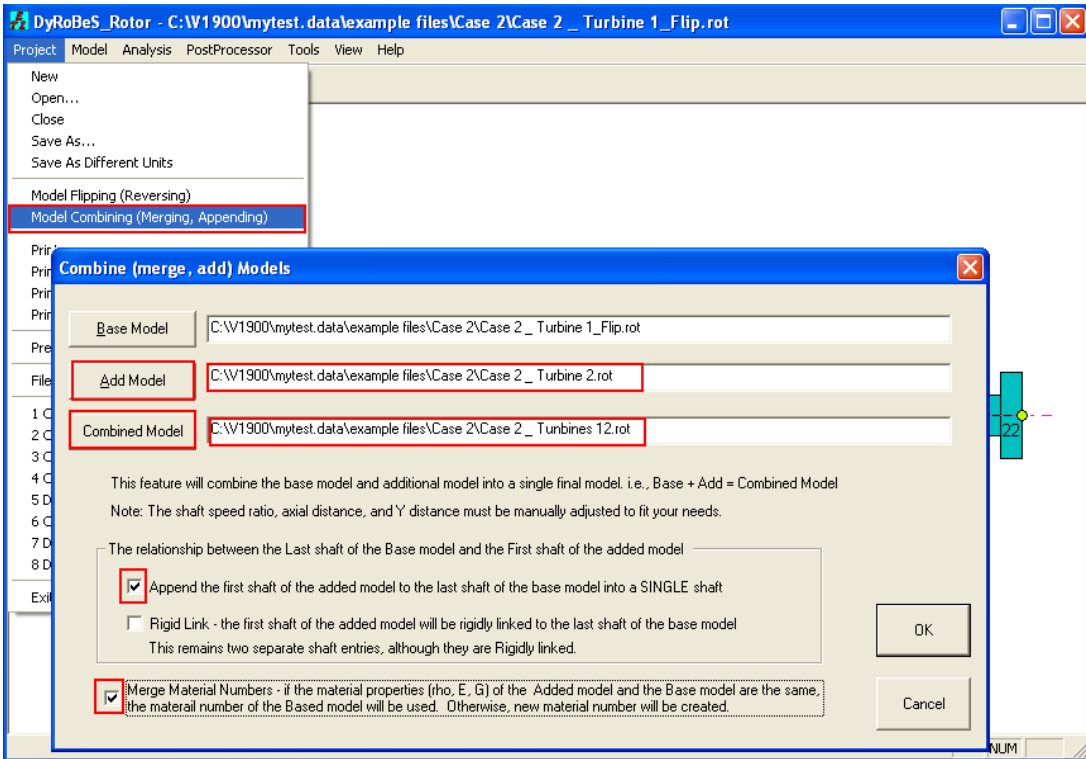


Step 1 is to flip the rotor 1 model before combining with others: Open the original rotor 1 file, Select Model Flipping under Project menu, then Save the flipped model in a different file name (in case, the original file is needed for other purposes). After this is done, the main window will be updated with this new file name and new model.



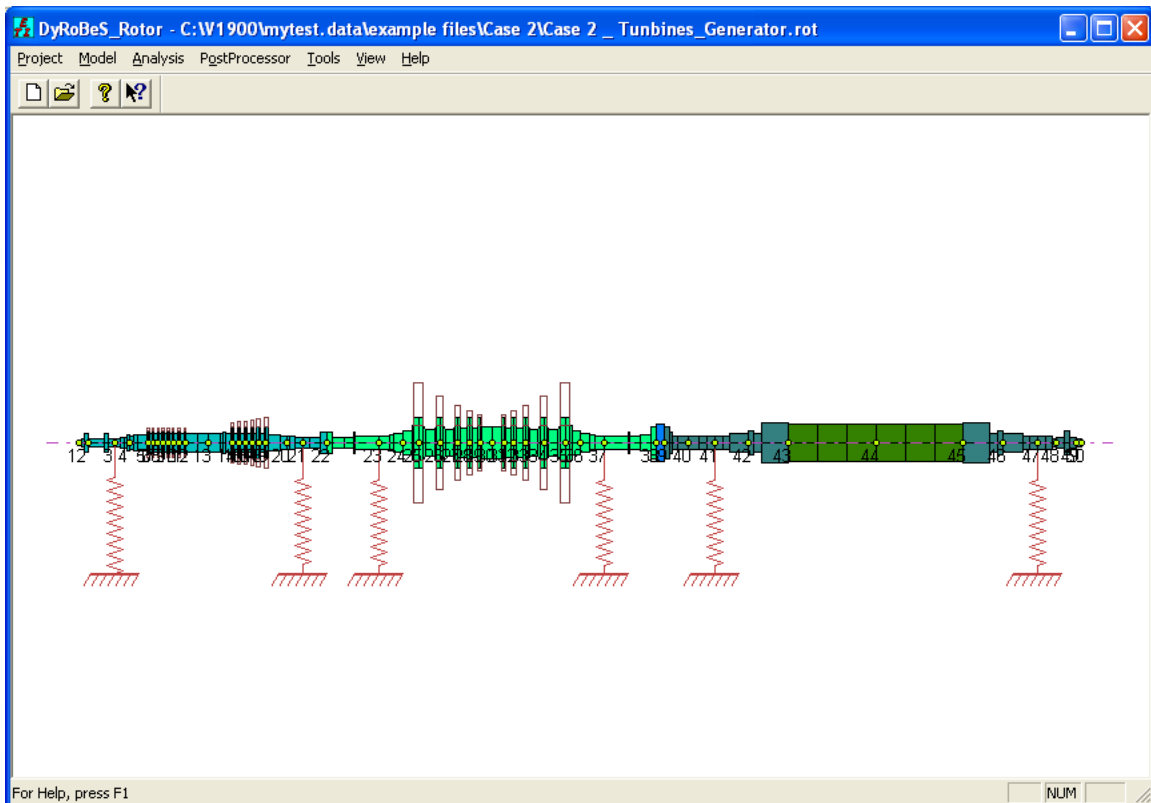
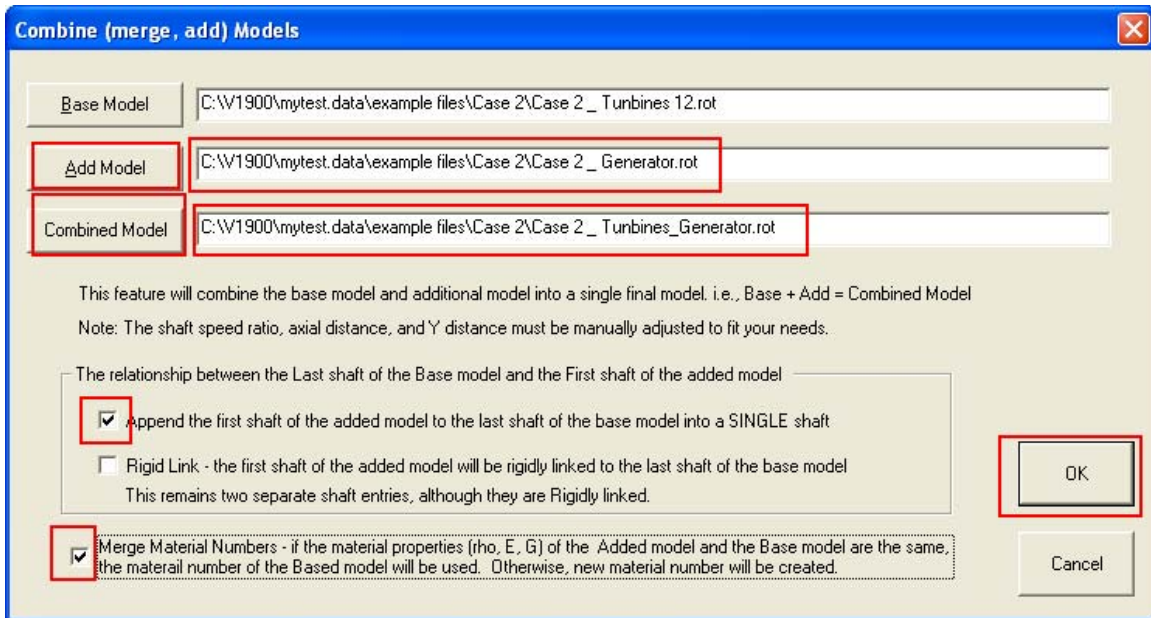


Step 2 to combine (merge, append) the Rotor 2 file into this rotor 1: Select Model Combining under Project menu while the Rotor 1 is still an active project in the screen. A dialog box will appear once the Model Combining is selected. Since the rotor 1 is still active and is used as the base model. Click Add Model to add the second rotor and also click Combined Model to save the combined model into a file. Since all three rotors are in-line in this case, we can simply append the rotor 2 into the rotor 1 in a single shaft without generating a second shaft in the model. We can also use Rigid Link option to keep the rotor 2 in a second shaft and rigidly link to Rotor 1. Both will yield the same analysis results. The Merge Material Number box is checked, that is, if the material properties (Rho, E, G, and description) in rotor 2 can be found in the Rotor 1 material library, then the existing material number in Rotor 1 will be used for Rotor 2, and it will not duplicate the same material in another material number. If the Merge Material box is not checked, the material number in rotor 2 will append to the material numbers in Rotor 1 as a new material. After this step, the main window will be updated with this new combined model.

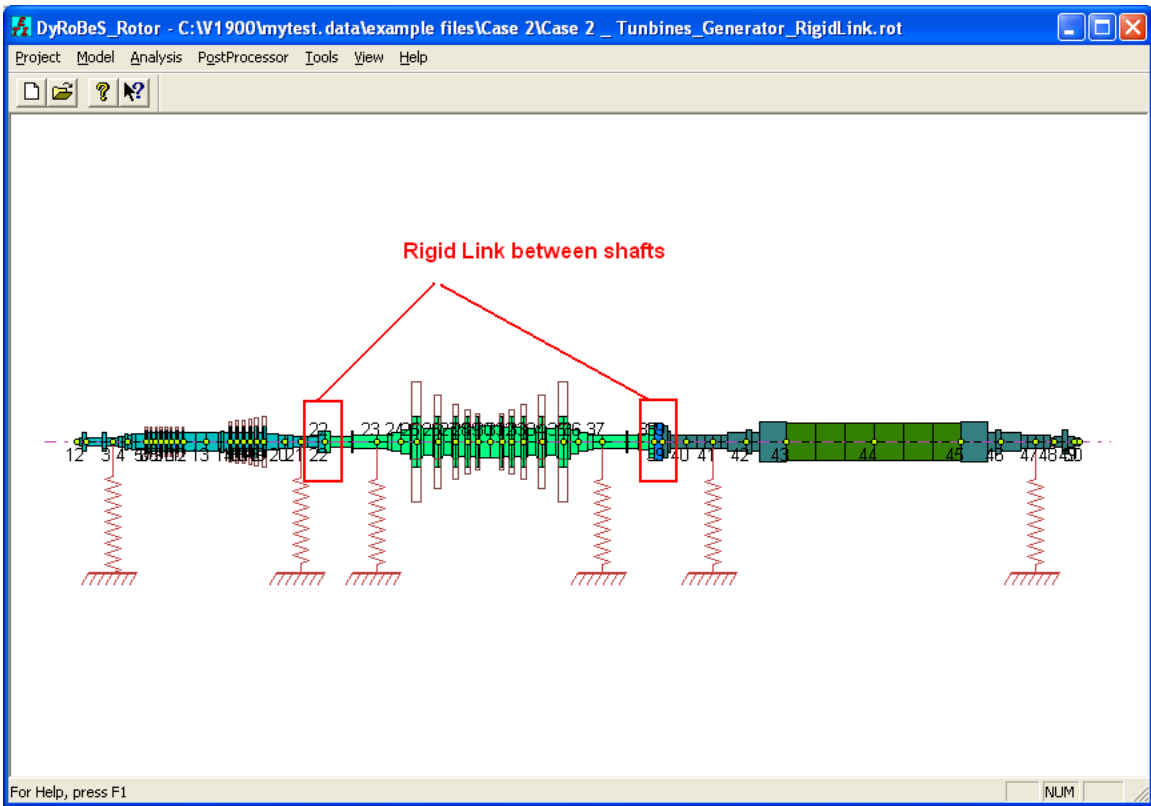
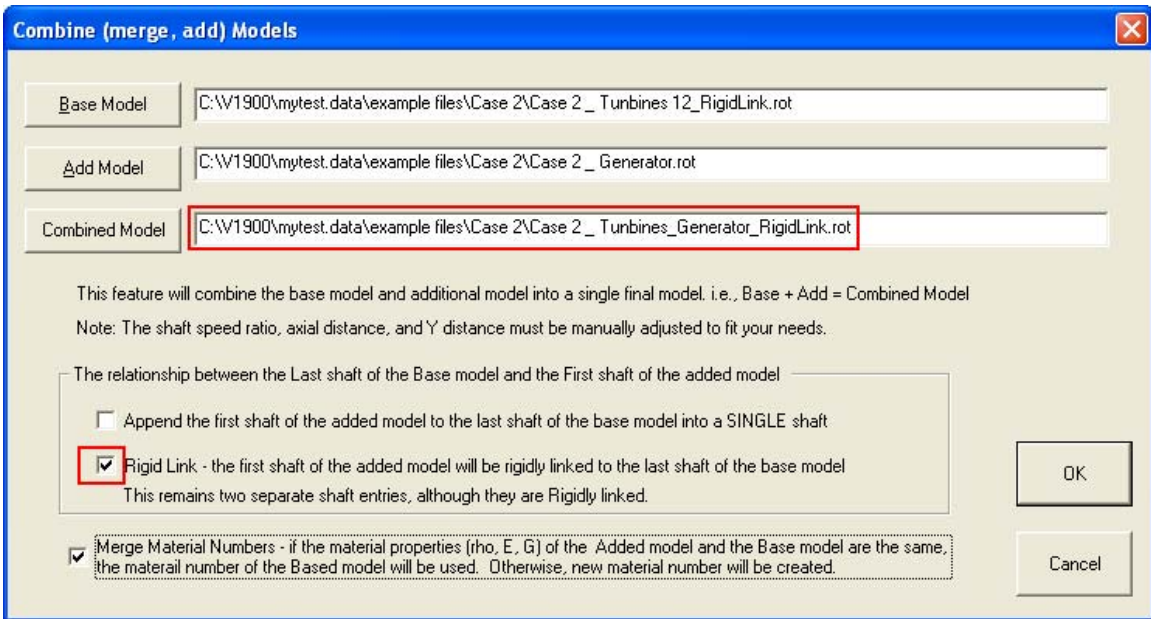


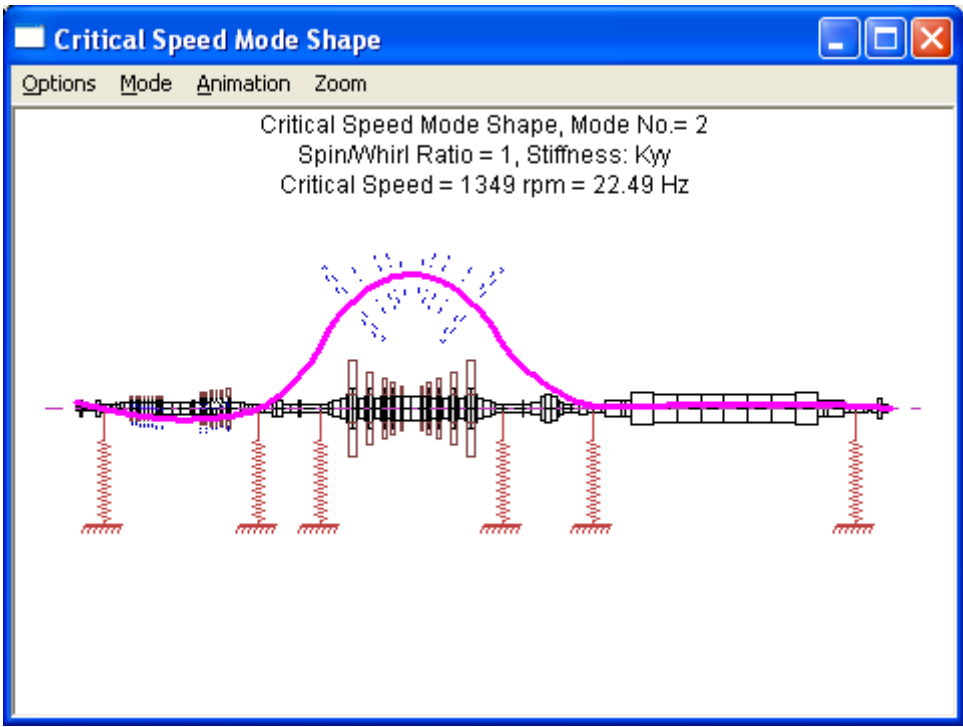
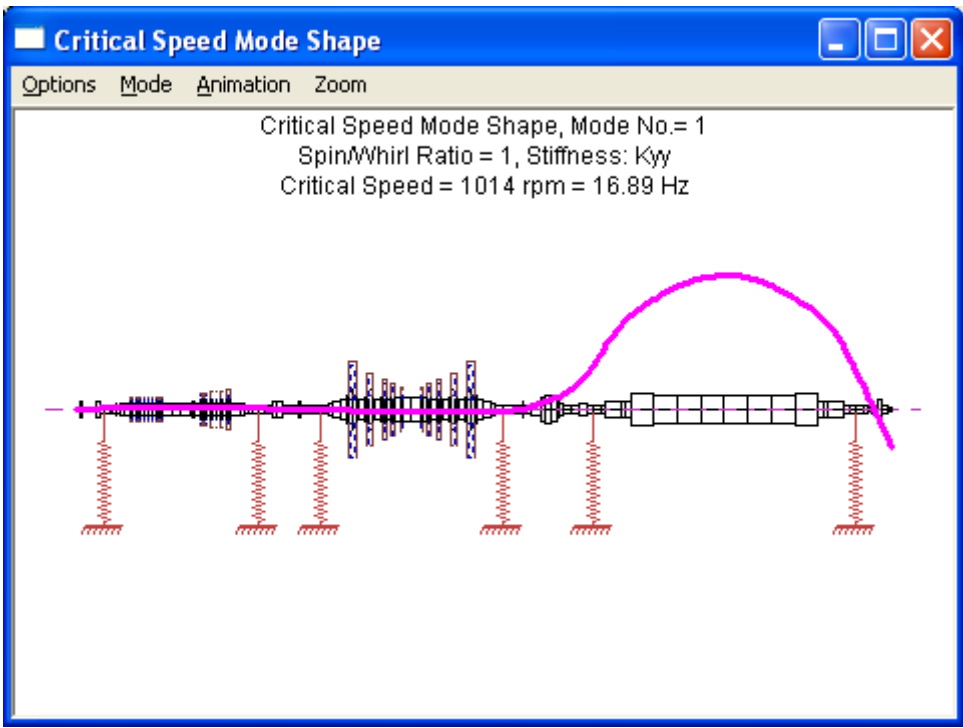
Step 3 is to combine the generator (rotor 3). Again, Select Model Combining under Project menu and enter the necessary information as shown below to add the generator

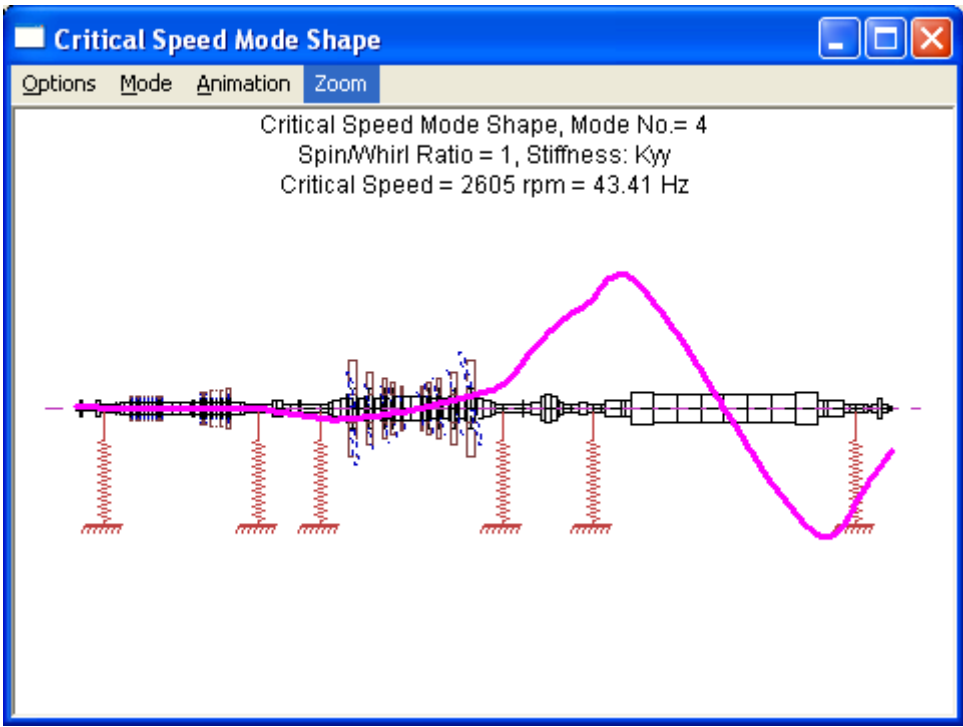
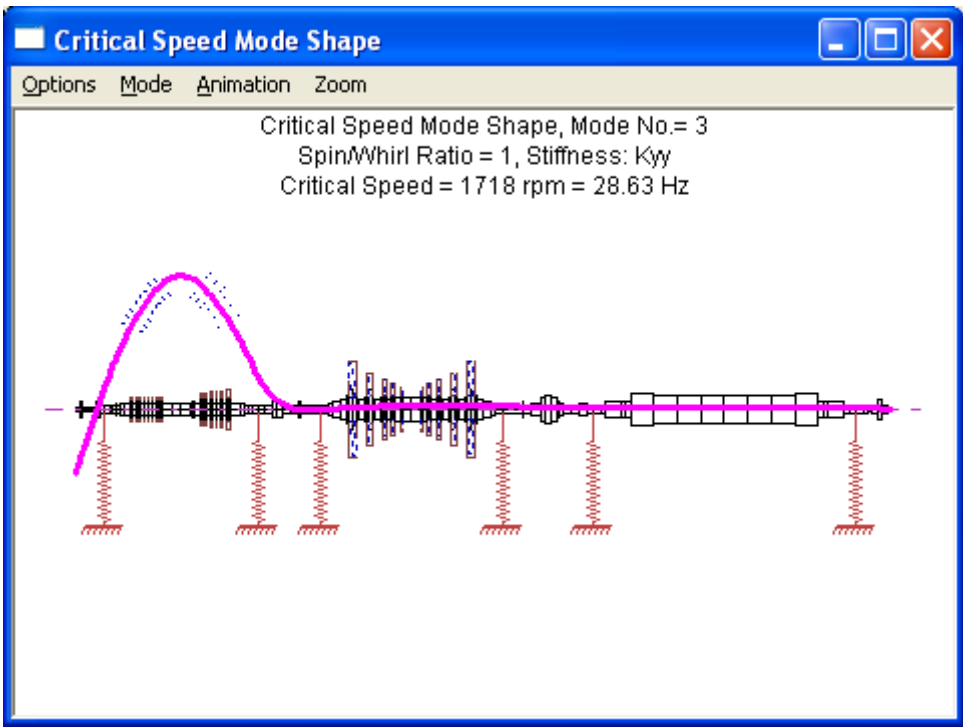
into the system. After these two steps, the separate three rotors are combined into a single model.



If you prefer to keep these three rotors in separate shafts in the system, then you may use the Rigid Link option instead of the Append in the check box as shown below. Both methods yield the same analysis results.

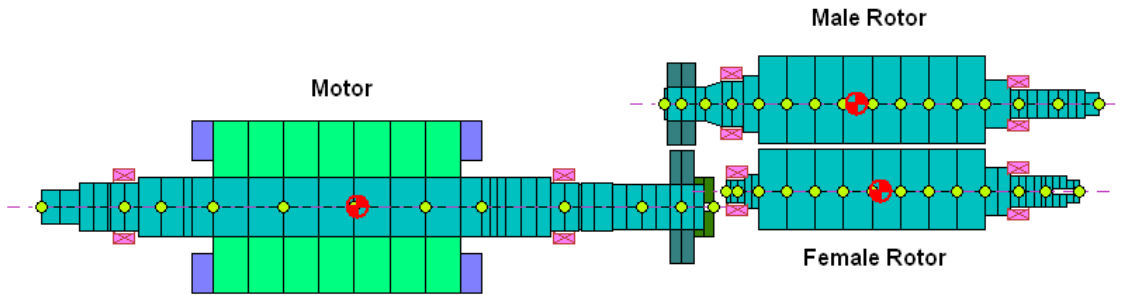






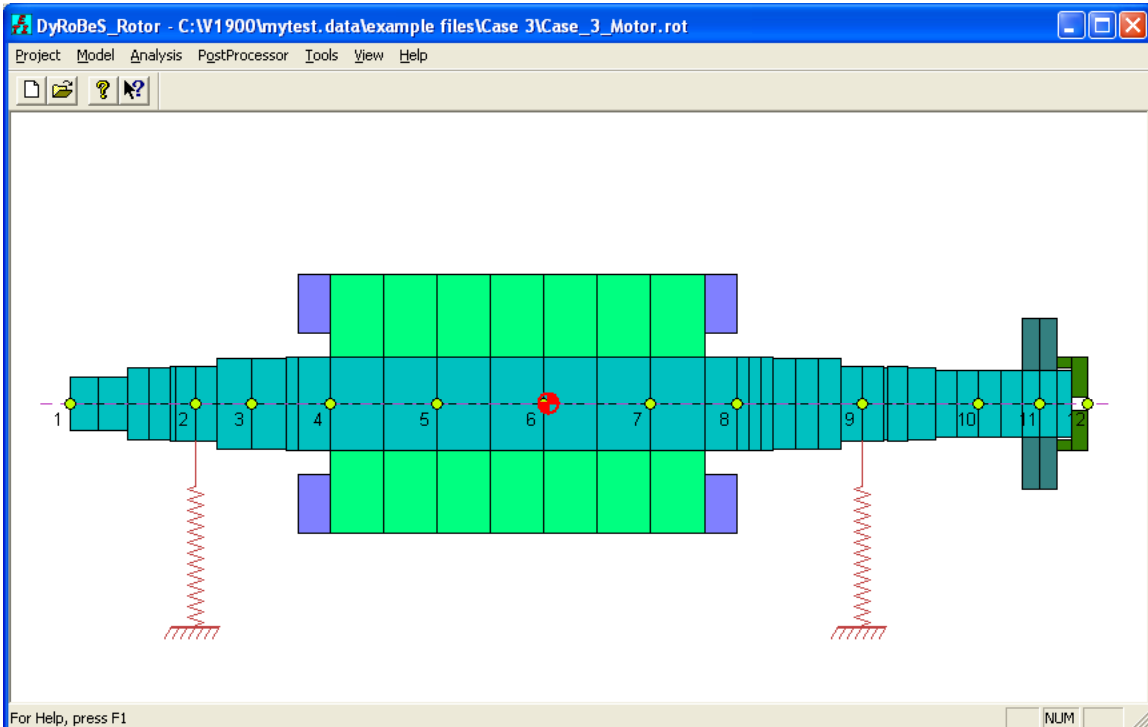
Case 3 – Rotary Compressor Train

In this example, we will demonstrate the coupled lateral and torsional analysis and the effect of the gear mesh coupling on the system natural frequencies. This system is a motor, which drives a rotary compressor with a male rotor and a female rotor.



Again, the rotors are modeled separately as shown below. Each rotor is analyzed first before assembly into a complete system. For each individual rotor, the **lateral** whirl speed analysis is performed to examine the system natural frequencies. They show that all the natural frequencies are far above the rotor operating speed and all potential excitation frequencies. Also, the gyroscopic effect is insignificant in this example. The natural frequencies do not vary much with the rotor speed for all three rotors. However, the coupled lateral and torsional natural frequencies of the entire system (3 rotors together) decreased significantly in this example show below.

Motor



DyRoBeS_Rotor - C:\W1900\mytest_data\example files\Case 3\Case_3_Motor_rot

Project Model Analysis PostProcessor Tools

Lateral Analysis Option & Run Time Data

Analysis: 4 - Whirl Speed & Stability Analysis

Shaft Element Effects

- Rotatory Inertia
- Shear Deformation
- Gyroscopic
- Gz

Static Deflection

- Constrained Bearing Stations

Critical Speed Analysis

Spin/Whirl Ratio: 1

No. of Modes: 3

Brg Stiffness: (Kxx+Kyy)/2

@ rpm: 0

Critical Speed Map

Spin/Whirl Ratio: 1

Bearing K - Min: 100

Npts: 50 Max: 1e+009

Stiffness to be varied at

Bearings: All

- Allow Bearings in Series

Transient Analysis

RPM: 0

Time Domain

Constant Speed: 0 rpm

Time-Start: 0

Ending: 0

Increment: 0

Solution Method

Wilson-theta

Initial Cs: No

- Mass
- Const
- Shaft
- Disk
- Gravi
- Gravi
- Static
- Time
- Misali

Whirl Speed and Stability Analysis

RPM-Starting: 0

Ending: 5000

Increment: 500

No. of Modes: 20

Steady State Synchronous Response Analysis

RPM-Starting: 0

Ending: 0

Increment: 0

Excitation Shaft: 1

- All Synchronized Shafts

Effects:

- Mass Unbalance
- Const. Unbalance
- Shaft Bow
- Disk Skew
- Misalignment

Steady State Harm

RPM-Starting: 0

Ending: 0

Increment: 0

Excitation Shaft: 1

- All Shafts with

Steady Maneuvers (Base Constant Translational Acceleration and/or Turn Rate)

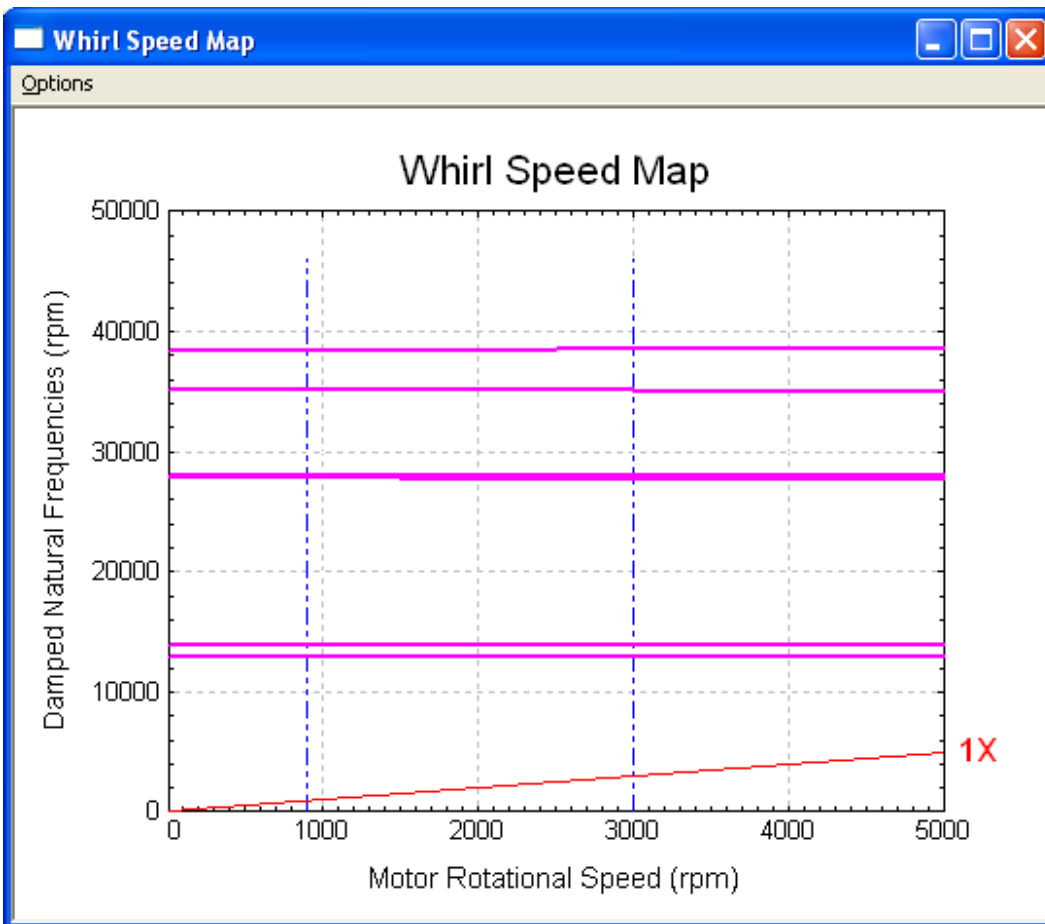
Speed (RPM): 0

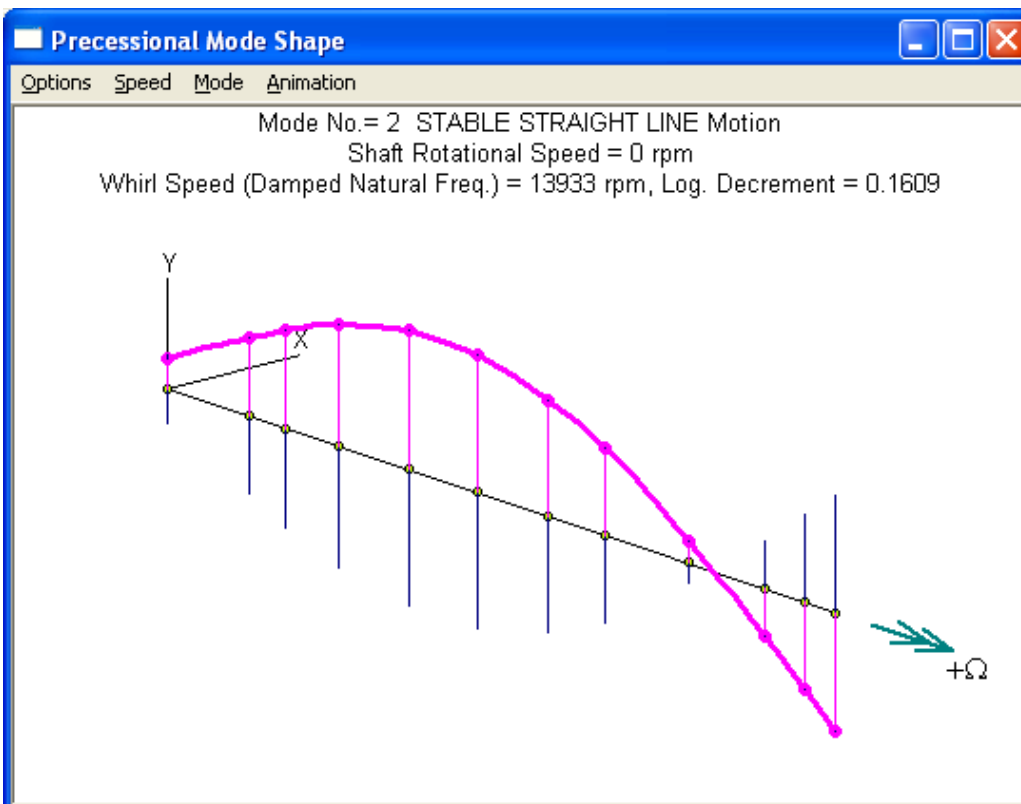
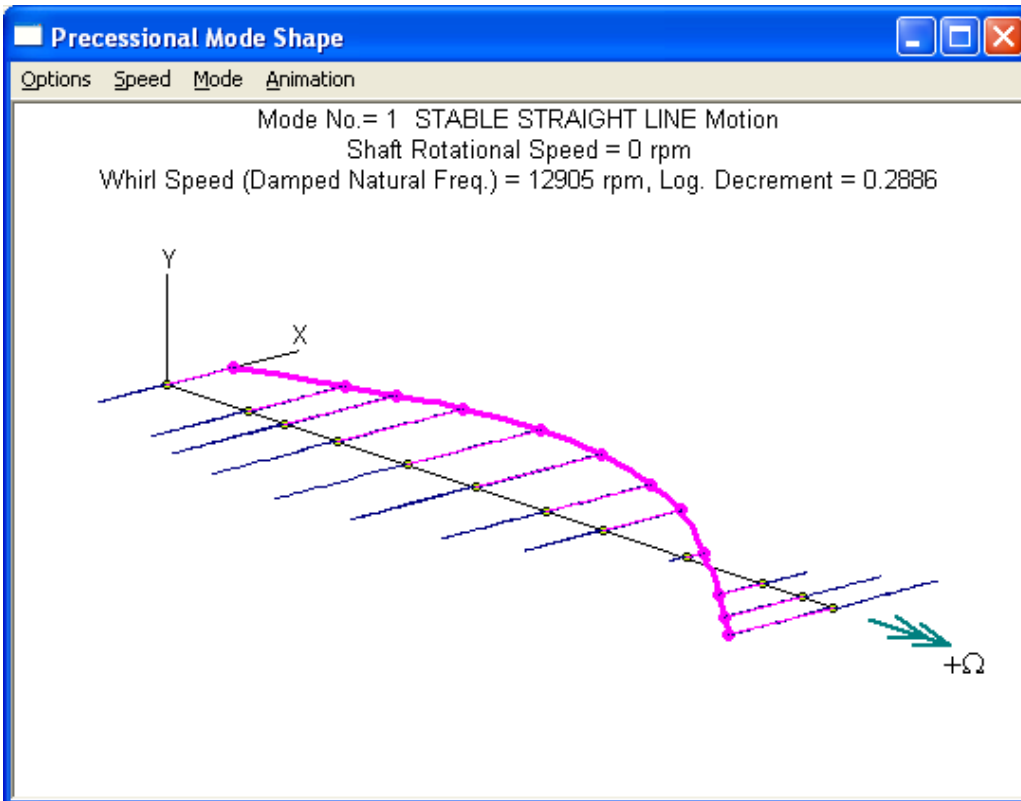
Acceleration - X: 0

Y: 0

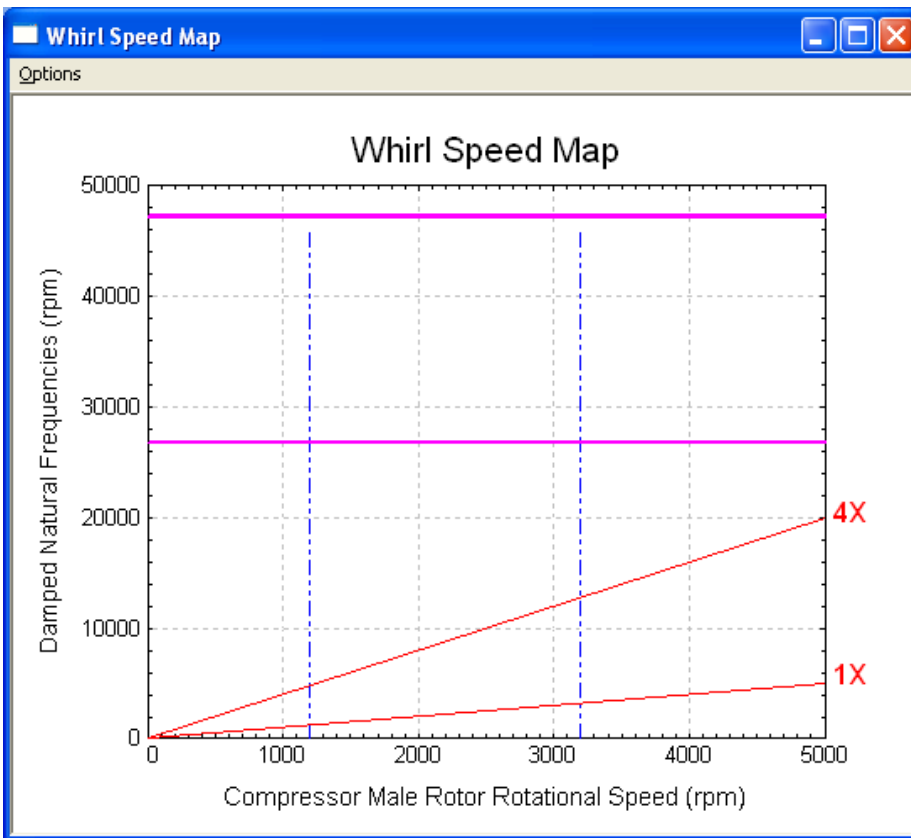
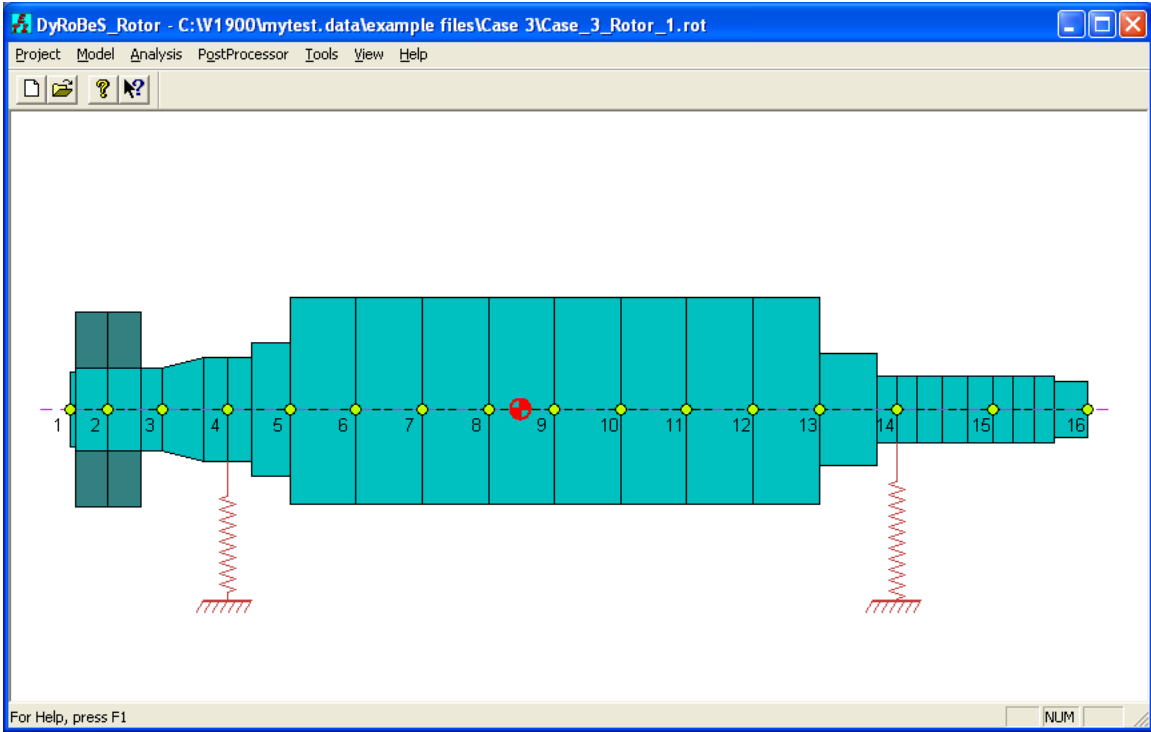
Turn Rate - X: 0

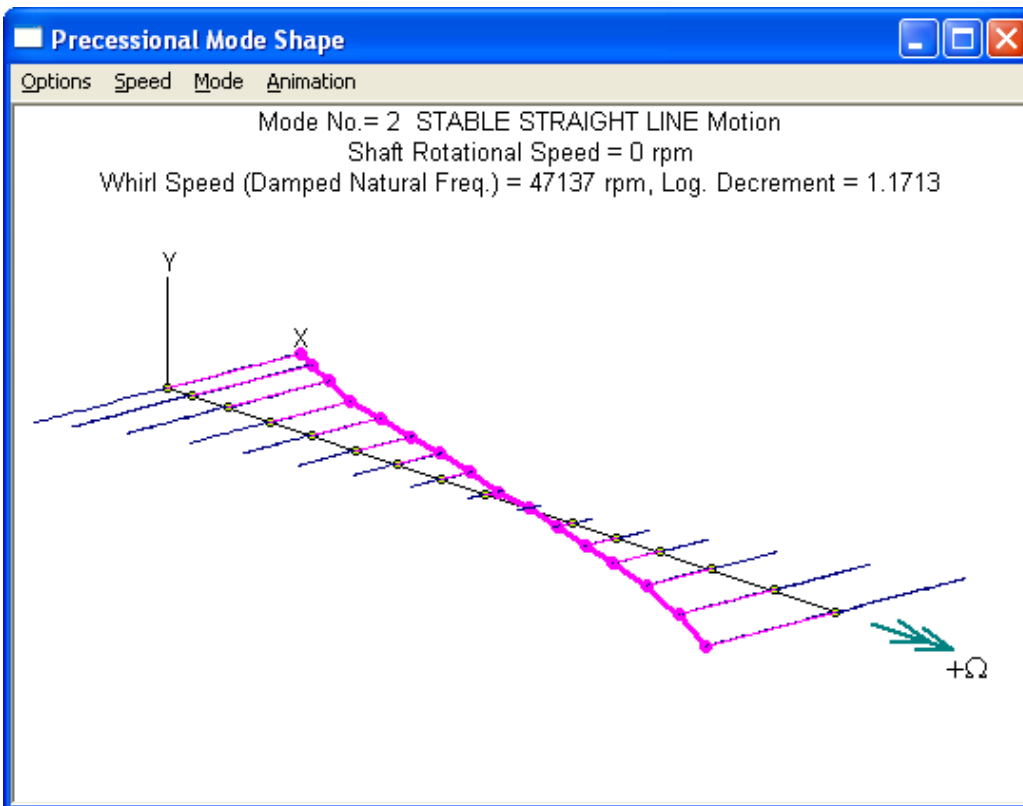
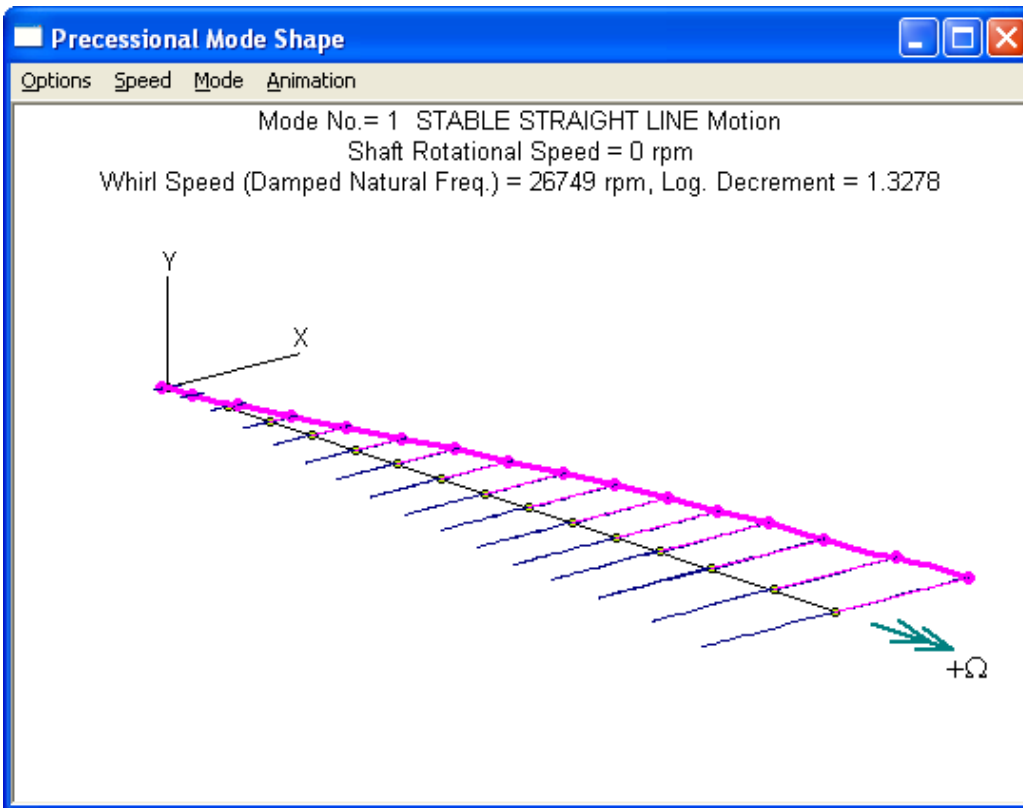
Y: 0



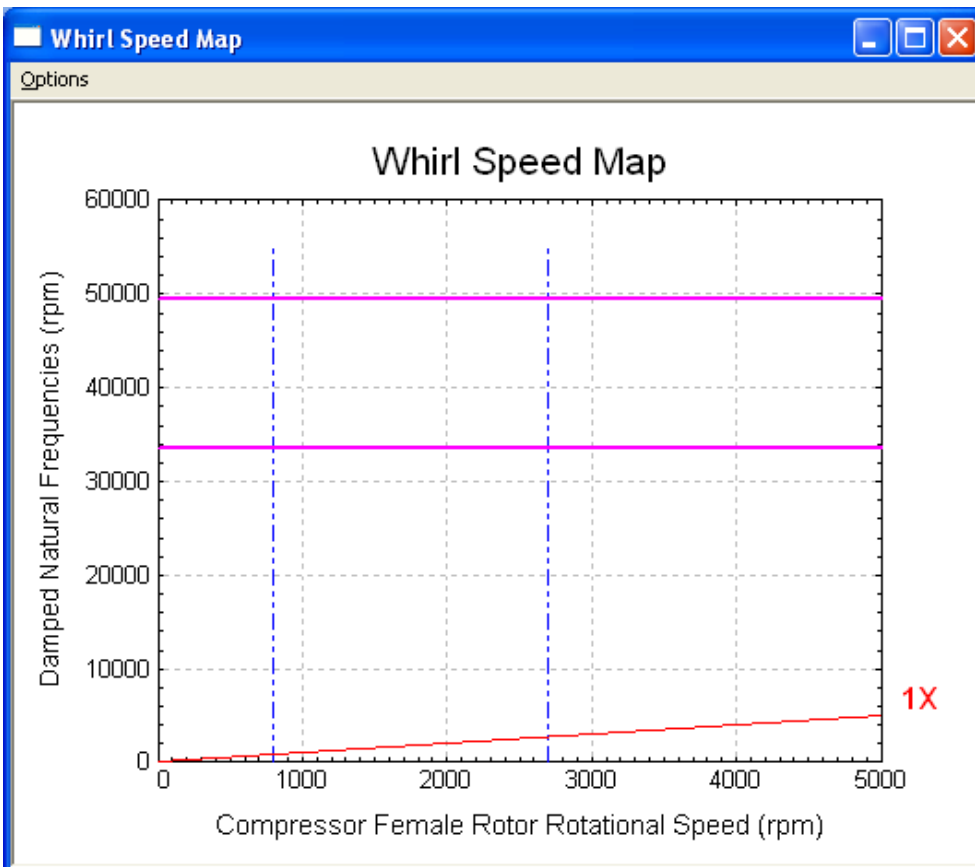
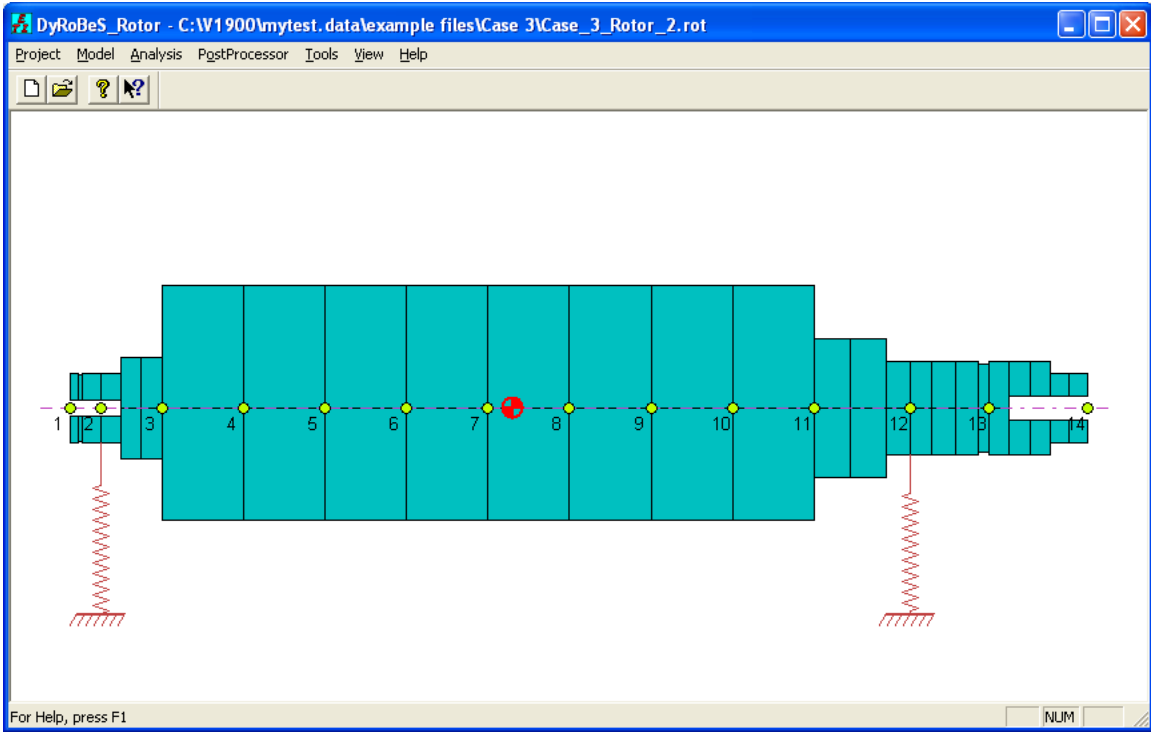


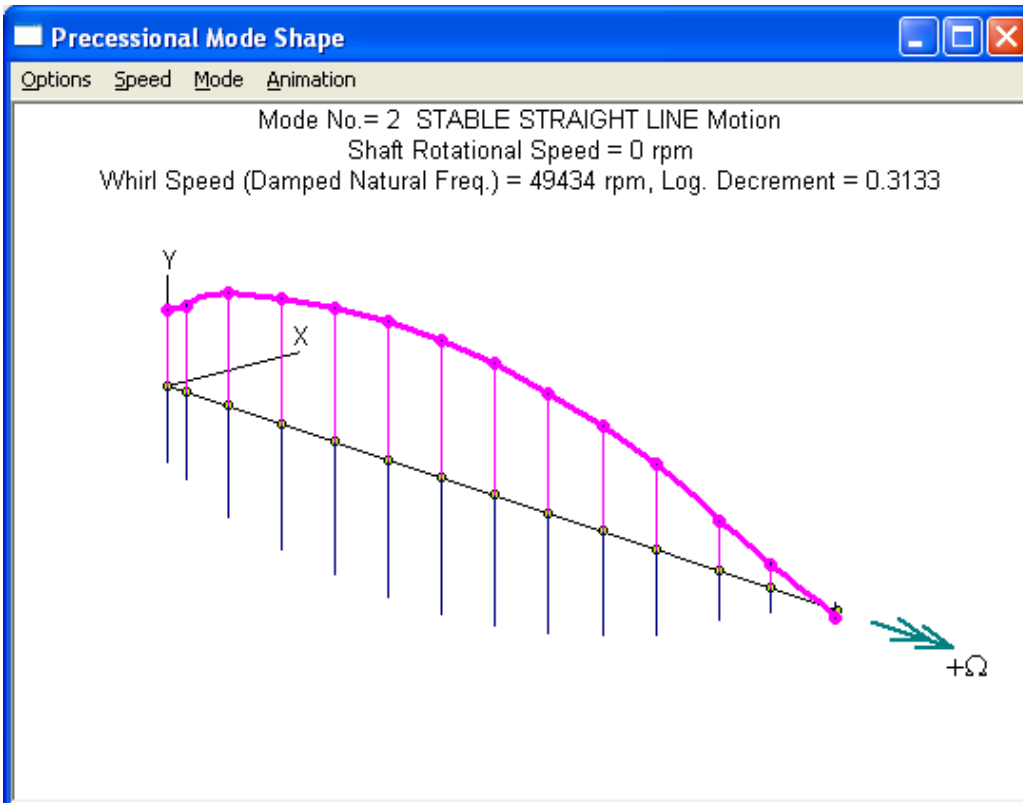
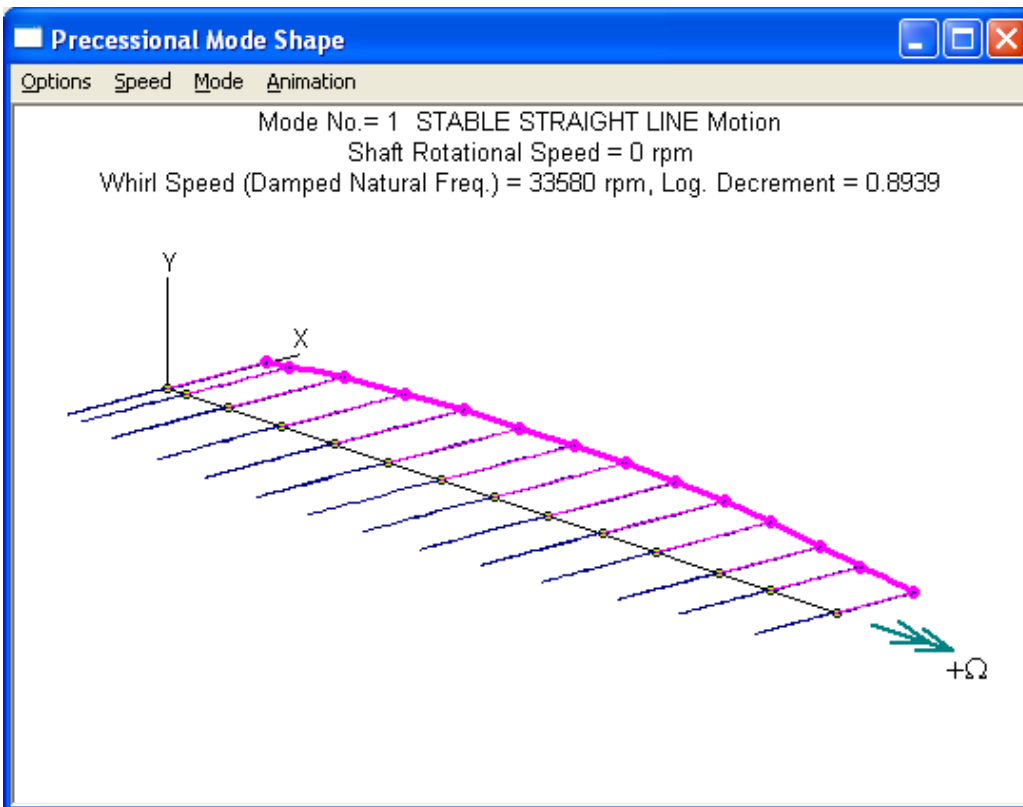
Compressor – Male Motor





Compressor – Female Rotor

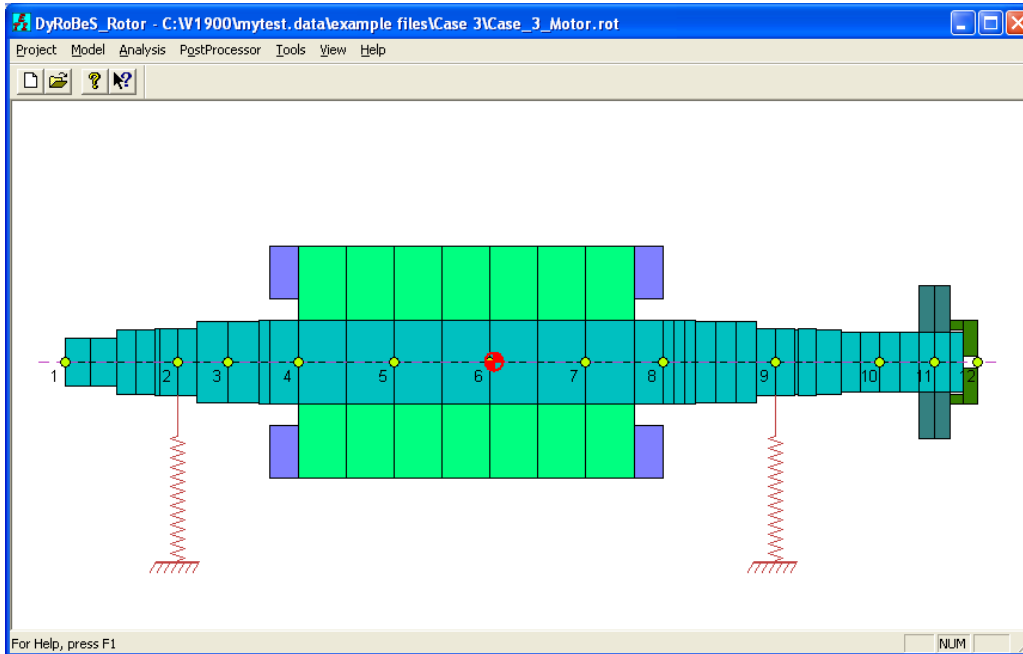




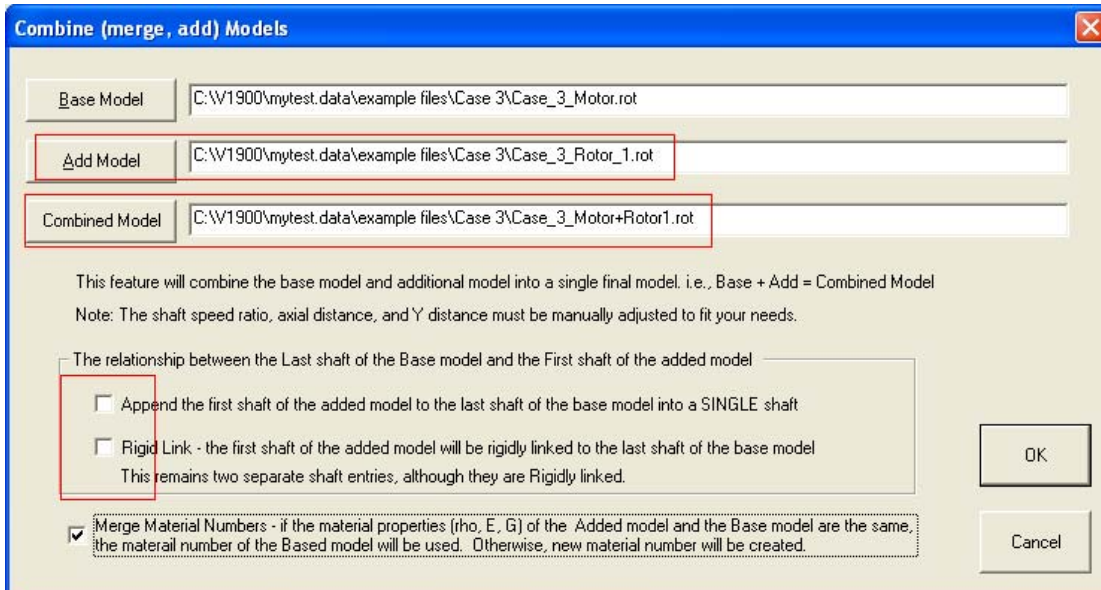
Combined Gear Train

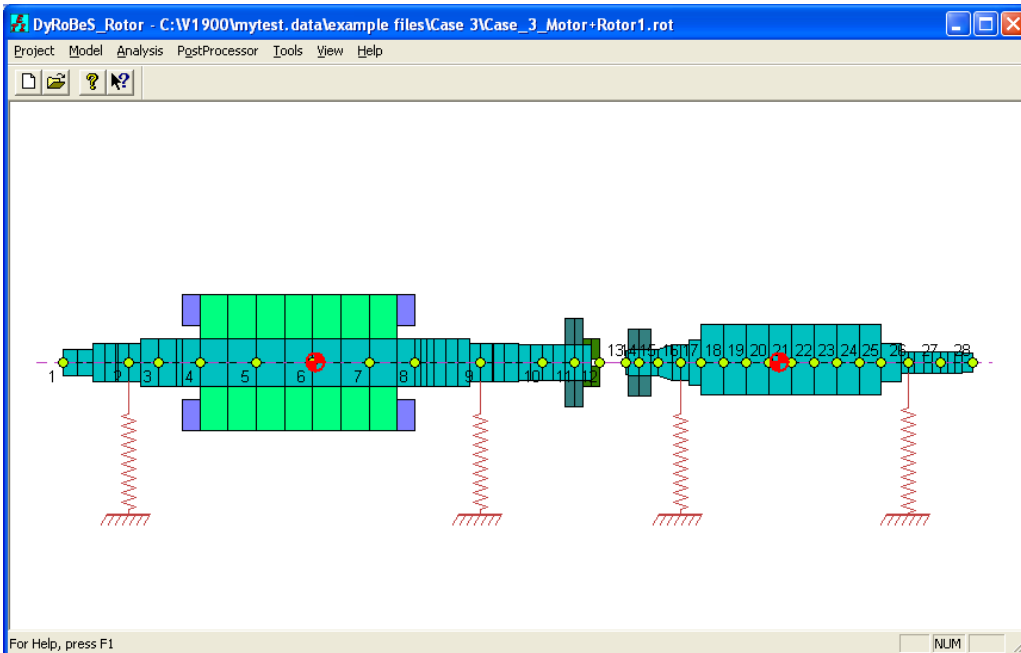
Now, let us combine three rotors together. The steps are described below:

1. Open the Motor Model as a base model



2. Select Model Combining under Project menu and enter the data blow. In this example, three rotors are coupled by the gear meshes. Therefore, do **NOT** check the Append and Rigid Link boxes. After this is done, the combined model will be shown in the main window. Since the Append and Rigid Link boxes are not checked, the second rotor will be shown in the right hand side of the first rotor with a small gap in between.





3. Since this is a geared system, we need to enter the speed ratio and also adjust the rotor position for proper display. Model – Data Editor – Shaft Elements – Second Shaft. Enter speed ratio, Axial and Y distances.

Rotor Bearing System Data

Axial Forces | Static Loads | Constraints | Misalignments | Shaft Bow | Time Forcing | Harmonics | Torsional/Axial
 Units/Description | Material | **Shaft Elements** | Disks | Unbalance | Bearings | Supports | Foundation | User's Elements

Shaft: 2 of 2 Starting Station #: 13 R Link Add Shaft Del Shaft Previous Next

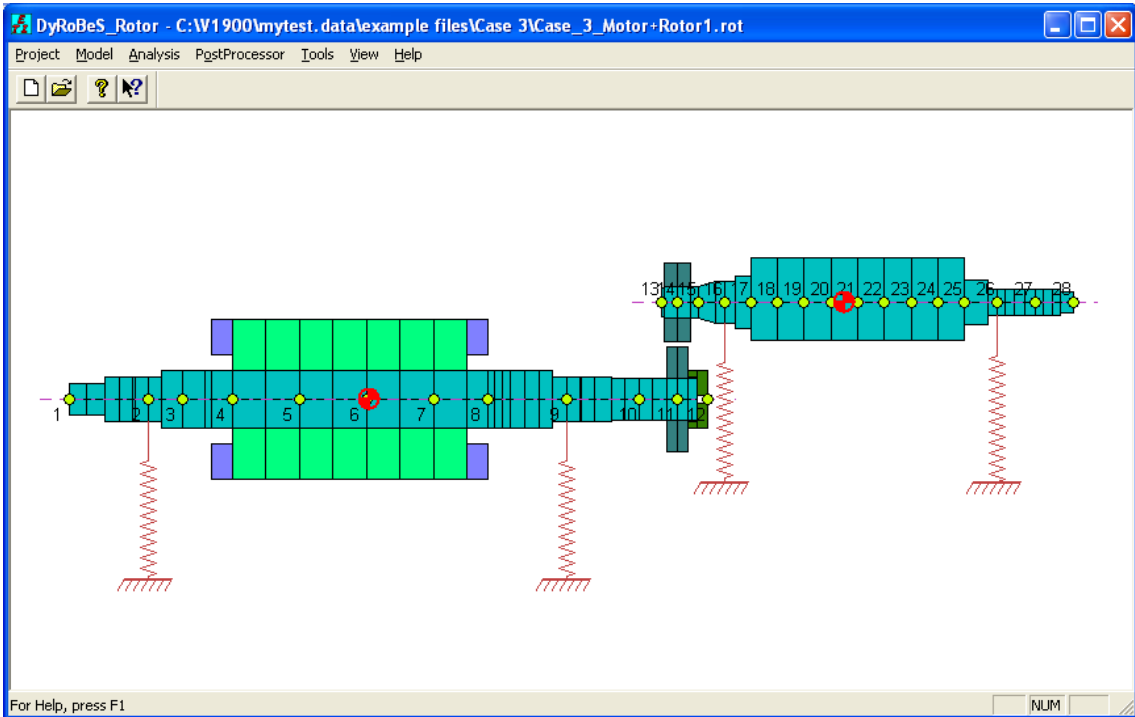
Speed Ratio: -1.347 Axial Distance: 757.529 Y Distance: 125 Import *.xls Export *.xls

Comment: Compressor Male Rotor

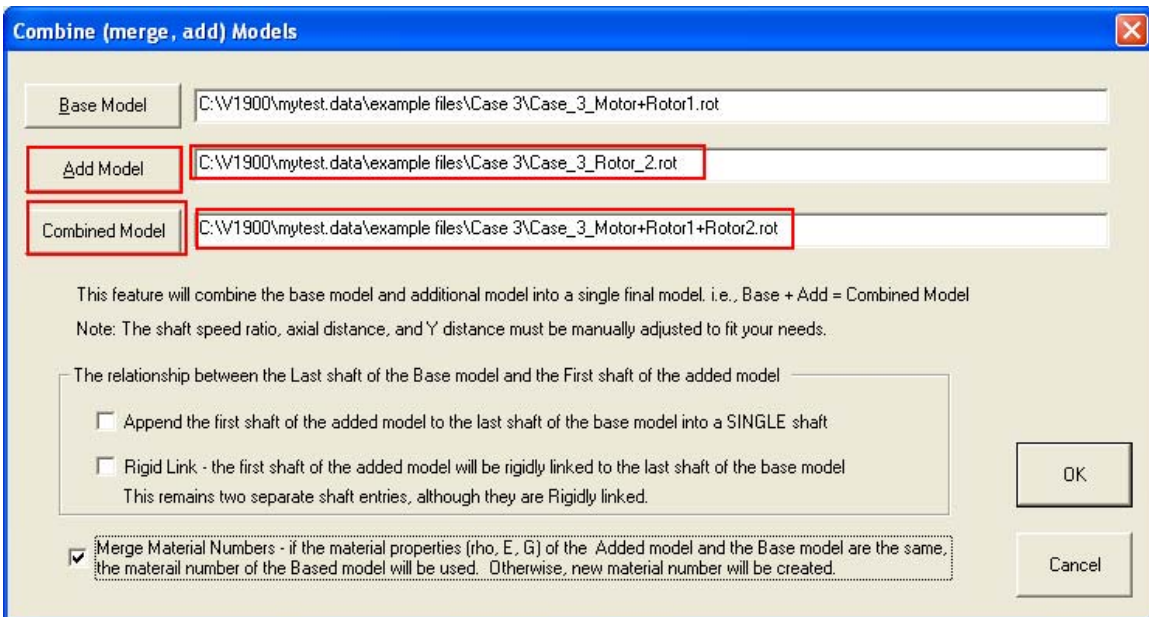
	Ele	Sub	Mat	Lev	Length	Mass ID	Mass OD	Stiff ID	Stiff OD	Comments
1	13	1	1	0	3	0	39	0	39	
2	13	2	1	0	17	0	43.5	0	43.5	
3	13	2	5	1	17	43.5	102.133	43.5	102.133	
4	14	1	1	0	17	0	43.5	0	43.5	
5	14	1	5	1	17	43.5	102.133	43.5	102.133	
6	14	2	1	0	11	0	43.5	0	43.5	
7	15	-1	1	0	21.2	0	43.5	0	55	
8	15	2	1	0	12.65	0	55	0	55	
9	16	1	1	0	12.65	0	55	0	55	
10	16	2	1	0	19.5	0	69.288	0	69.288	
11	17	1	1	0	34.25	0	116.2	0	98	
12	18	1	1	0	34.25	0	116.2	0	98	
13	19	1	1	0	34.25	0	116.2	0	98	
14	20	1	1	0	34.25	0	116.2	0	98	
15	21	1	1	0	34.25	0	116.2	0	98	
16	22	1	1	0	34.25	0	116.2	0	98	
17	23	1	1	0	34.25	0	116.2	0	98	
18	24	1	1	0	34.25	0	116.2	0	98	
19	25	1	1	0	30.21	0	59.288	0	59.288	
20	25	2	1	0	10.25	0	35.015	0	35.015	

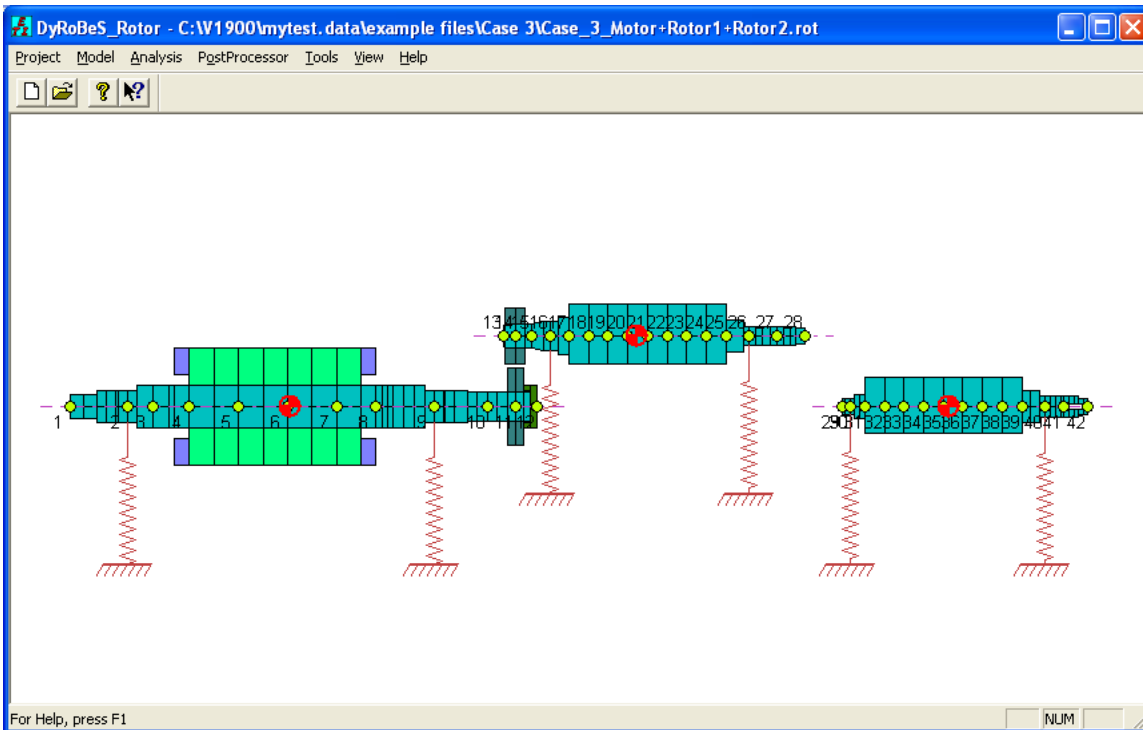
Unit:(4) - Length, Diameter: mm

Insert Row Delete Row ReNumber Copy & Paste Save Save As Close Help



4. Combine the last rotor into the system. Again, we need to enter the speed ratio and the axial and Y distances in the Shaft Elements tab. We also need to enter the gear mesh data to couple these three rotors.





Rotor Bearing System Data

Axial Forces | Static Loads | Constraints | Misalignments | Shaft Bow | Time Forcing | Harmonics | Torsional/Axial
 Units/Description | Material | Shaft Elements | Disks | Unbalance | Bearings | Supports | Foundation | User's Elements

Shaft: 3 of 3 Starting Station #: 29 R Link Add Shaft Del Shaft Previous Next

Speed Ratio: 0.898 Axial Distance: 832.529 Y Distance: 20 Import *.xls Export *.xls

Comment: Compressor - Female Rotor

	Ele	Sub	Mat	Lev	Length	Mass ID	Mass OD	Stiff ID	Stiff OD	Comments
	1	29	1	1	0	3.6	6.8	30.012	6.8	30.012
	2	29	2	1	0	1.505	6.8	28.4	6.8	28.4
	3	29	3	1	0	8.1975	6.8	30.012	6.8	30.012
	4	30	1	1	0	8.1975	6.8	30.012	6.8	30.012
	5	30	2	1	0	8.75	0	43.001	0	43.001
	6	30	3	1	0	8.75	0	43.001	0	43.001
	7	31	1	1	0	34.25	0	104	0	95
	8	32	1	1	0	34.25	0	104	0	95
	9	33	1	1	0	34.25	0	104	0	95
	10	34	1	1	0	34.25	0	104	0	95
	11	35	1	1	0	34.25	0	104	0	95
	12	36	1	1	0	34.25	0	104	0	95
	13	37	1	1	0	34.25	0	104	0	95
	14	38	1	1	0	34.25	0	104	0	95
	15	39	1	1	0	15.105	0	59.288	0	59.288
	16	39	2	1	0	15.105	0	59.288	0	59.288
	17	39	3	1	0	10.6075	0	40.015	0	40.015
	18	40	1	1	0	8.9375	0	40.015	0	40.015
	19	40	2	1	0	9.7725	0	40.015	0	40.015 brg
	20	40	3	1	0	9.7725	0	40.015	0	40.015

Unit:(4) - Length, Diameter: mm

Insert Row Delete Row ReNumber Copy & Paste Save Save As Close Help

Rotor Bearing System Data

Units/Description	Material	Shaft Elements	Disks	Unbalance	Bearings	Supports	Foundation	User's Elements
Axial Forces	Static Loads	Constraints	Misalignments	Shaft Bow	Time Forcing	Harmonics	Torsional/Axial	

Torsional and/or Axial Data

Linear Connectivity

Non-Linear Couplings/Connections

Modal Damping

Steady State Excitation (Single Harmonic)

Torsional Time Dependent Excitations

Excitations in Equations

Excitations from Files

I.C.

Torsional Startup Transient Torques

Motor Driving Torque

Load Torque

Reciprocating Excitation (nX Harmonics)

Engine Excitation

Recip. Torque

Lateral-Torsional-Axial Geared Coupling

Gear Mesh Data

Save Save As Close Help

Lateral-Torsional-Axial Gear Mesh Coupling

Gear Mesh: 1 of 2

Add Delete Previous Next OK

Gear 1

Station I: 11 Pitch Diameter: 137.57 Driving Driven Gear

Gear 2

Station J: 14 Pitch Diameter: 102.133 Angular Position: 180

Gear Data

Pressure Angle: 20

Helix Angle: 29.49

Gear Mesh Stiffness and Damping Matrices in (r't'a) coordinates

Stiffness				Damping			
K	r'	t'	a'	C	r'	t'	a'
r'	0	0	0	r'	0	0	0
t'	0	388865	0	t'	0	100	0
a'	0	0	0	a'	0	0	0

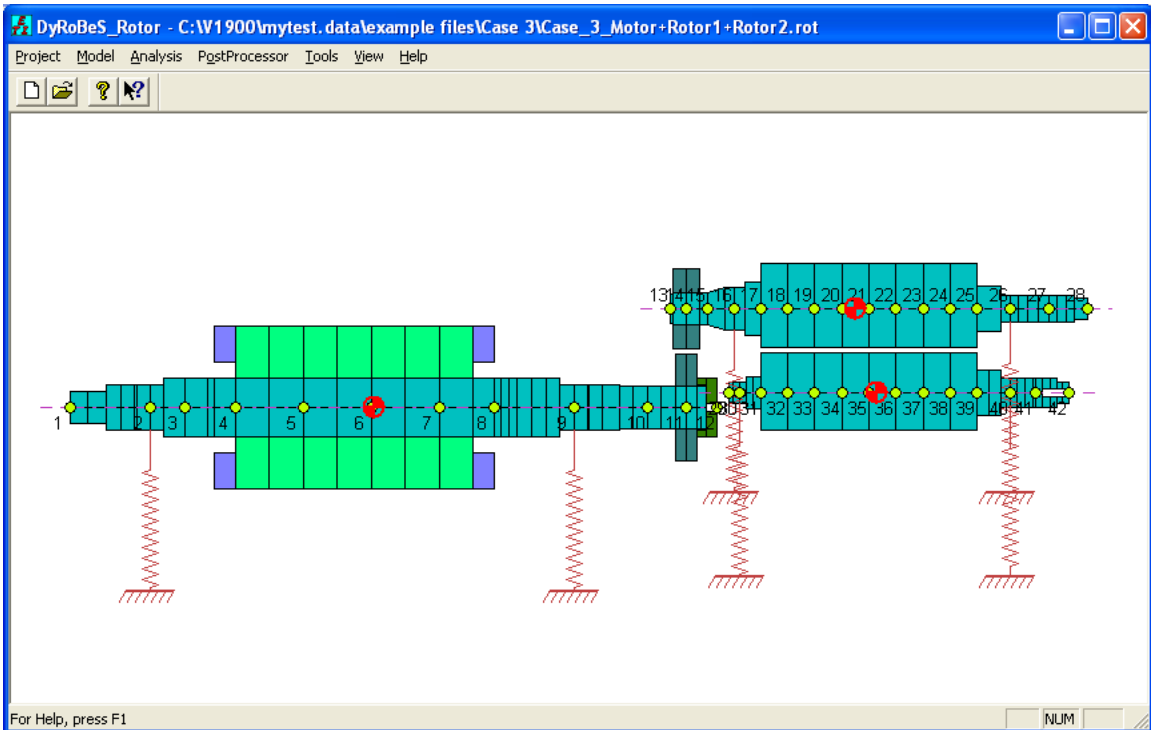
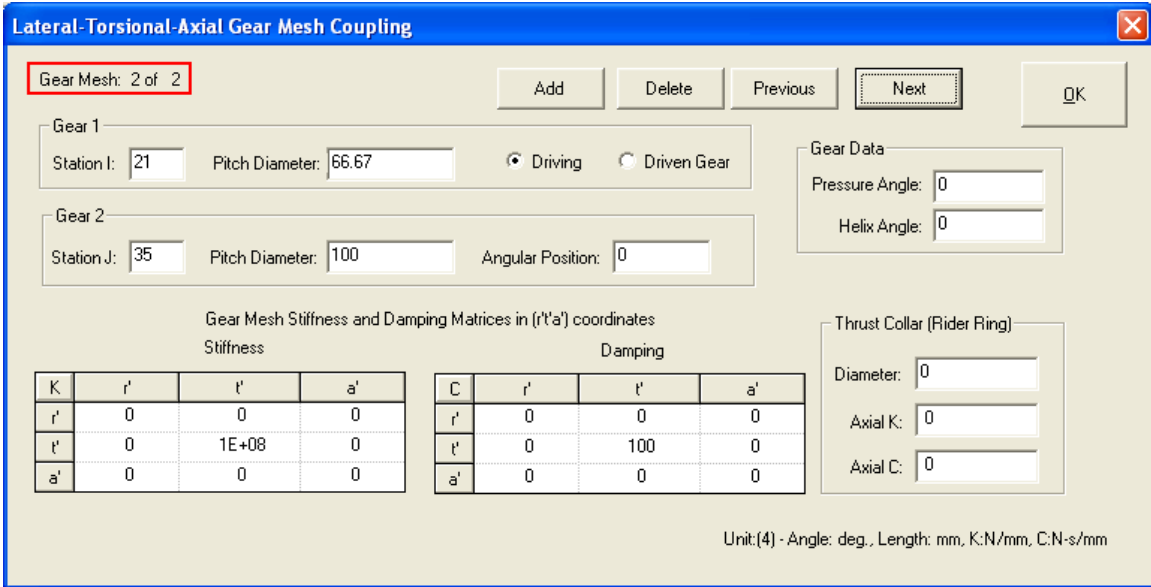
Thrust Collar (Rider Ring)

Diameter: 0

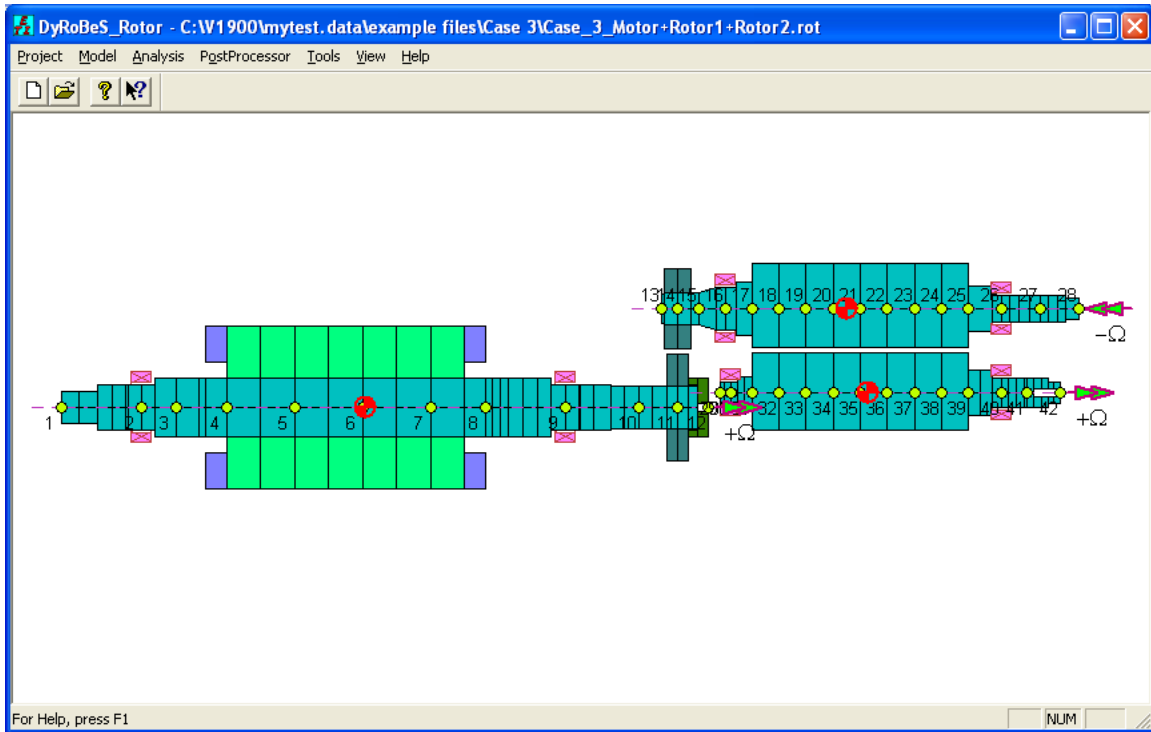
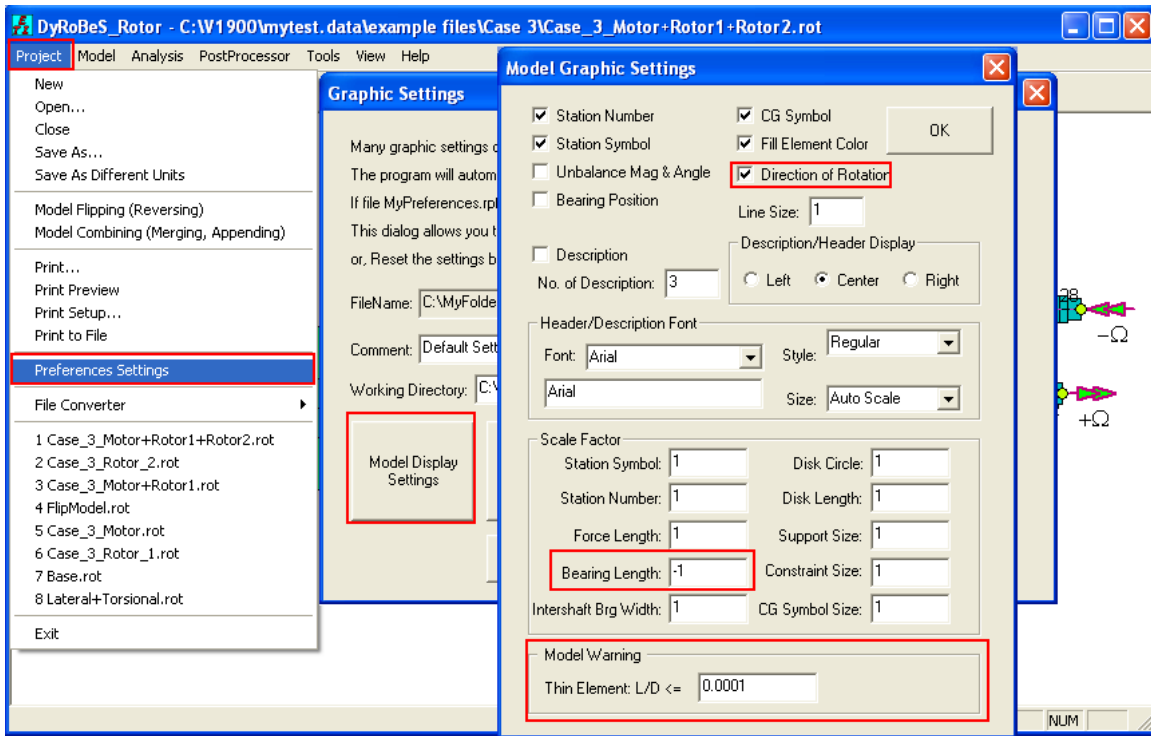
Axial K: 0

Axial C: 0

Unit:(4) - Angle: deg., Length: mm, K:N/mm, C:N-s/mm

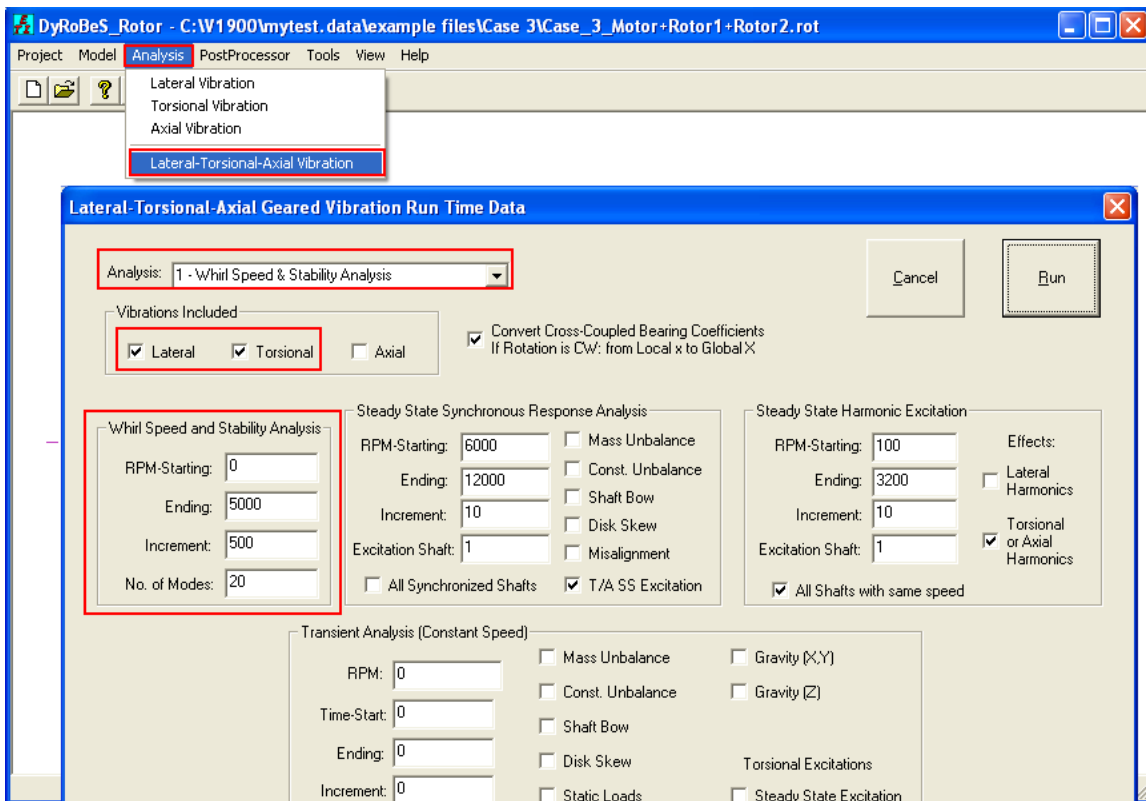


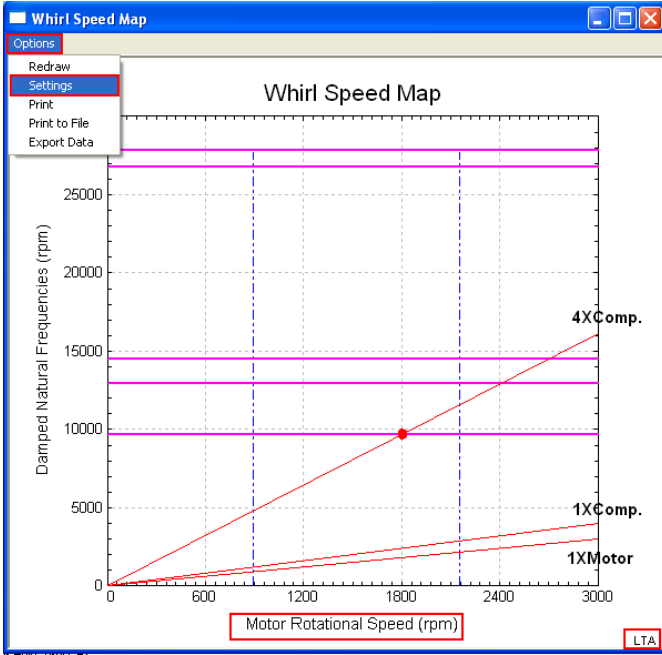
5. The regular bearing with a spring display can complicate the plot. We can display the bearing with a representing small box by changing the bearing length parameter under Project – Preference Settings – Model Display Settings – Bearing Length – make it a NEGATIVE value. Also, noted that one parameter is added in Ver 19 is the L/D ratio. If the element L/D ratio is less than this pre-defined value, a warning message will be prompted.



To perform the coupled lateral and torsional analysis, Select Lateral-Torsional-Axial Vibration under the Analysis menu. A dialog box shows up for the analysis inputs. Select Whirl Speed & Stability Analysis. In the Vibration Included group box, check the Lateral and Torsional boxes to include the lateral and torsional vibrations. That is, coupled lateral and torsional vibration through the gear meshes. Check the Convert

Cross-Coupled Bearing Coefficients box. This indicates that if the shaft rotates clockwise, then the bearing coefficients for this shaft will be converted to comply with this direction of rotation. For more details on this conversion, see book “Practical Rotordynamics and Fluid Film Bearing Design”, Chapter 8. For a multiple rotor system, the speed input in the analysis dialog is the first shaft speed. The first shaft in DyRoBeS is always the reference shaft. In this example, it is the motor speed. The compressor male rotor is designed with the speed range of 1200-2900 rpm. Therefore, the motor speed range is 891-2153 rpm. The excitations considered are 1X Motor, 1X Compressor (1.347X Motor), and 4X Compressor (5.388X Motor). In this example, there is one interference point at 9660 cpm present within the operating speed range. The X-Axis (speed axis) in the whirl speed map can be converted into the compressor male rotor speed by multiplying the X-Axis by the speed ratio of 1.347 in the Options-Settings. Once the X-axis is converted into the compressor male rotor speed, the 1X Motor excitation is 0.7424 X Compressor excitation. To avoid confusion, a small label “LTA” is shown in the right bottom of the plot to indicate this plot is from the coupled vibration option. The mode shape for this interference point at 9660 cpm frequency is also shown. For the coupled vibration, since the displacements are normalized with different units, different scale may be required. In this example, the torsional displacement is enlarged fifty times. Again, go to Options – Settings to make necessary adjustments.





Whirl Speed Map

Title: Whirl Speed Map

X-Label: Motor Rotational Speed (rpm)

Y-Label: Damped Natural Frequencies (rpm)

Number of Modes: 10

Excitation Slopes: 1, 1.347, 5.388 (r²) - For example: 0.5, 1, 2, 15

Operating Speed Range: 891 to 2153

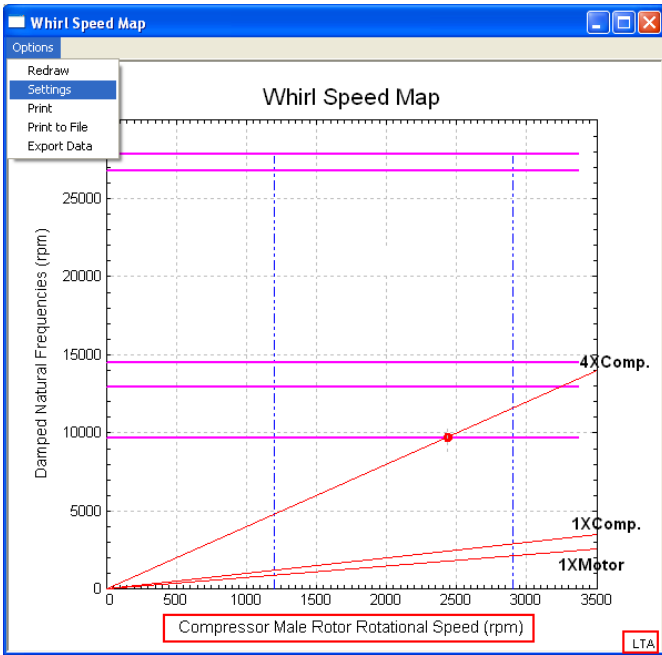
Safe Margin Speeds: 0 0

Curve Fit Symbol Major Grid Minor Grid

Manual Scaling

Manual Scaling Data:

Xmin: 0	Ymin: 0	X Scale: 1
Xmax: 3000	Ymax: 30000	Y Scale: 1
XDiv: 5	YDiv: 6	



Whirl Speed Map

Title: Whirl Speed Map

X-Label: Compressor Male Rotor Rotational Speed (rpm)

Y-Label: Damped Natural Frequencies (rpm)

Number of Modes: 10

Excitation Slopes: 0.7424, 1, 4 (r²) - For example: 0.5, 1, 2, 15

Operating Speed Range: 1200 to 2900

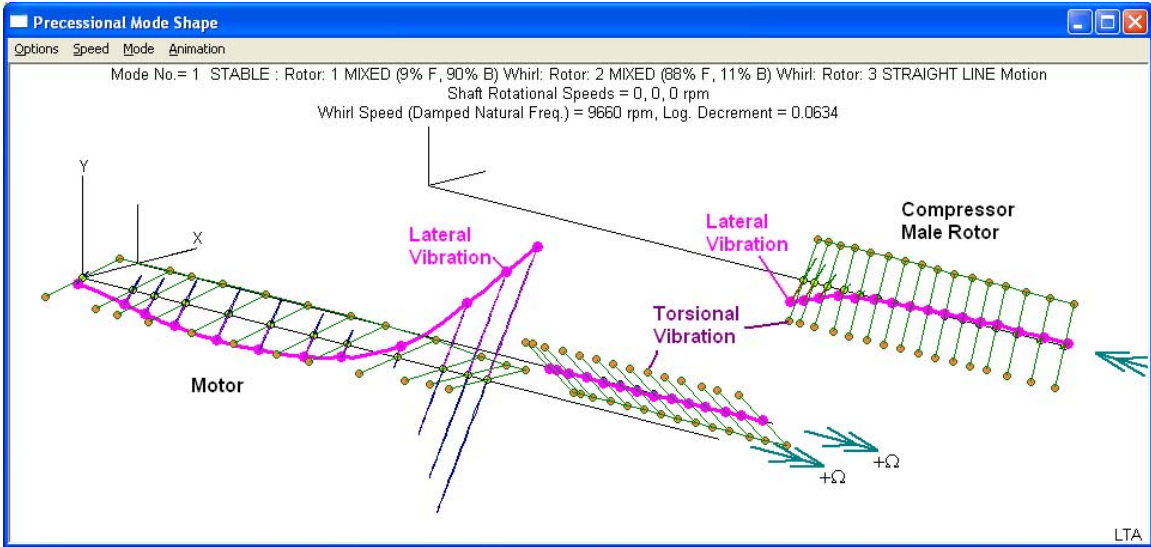
Safe Margin Speeds: 0 0

Curve Fit Symbol Major Grid Minor Grid

Manual Scaling

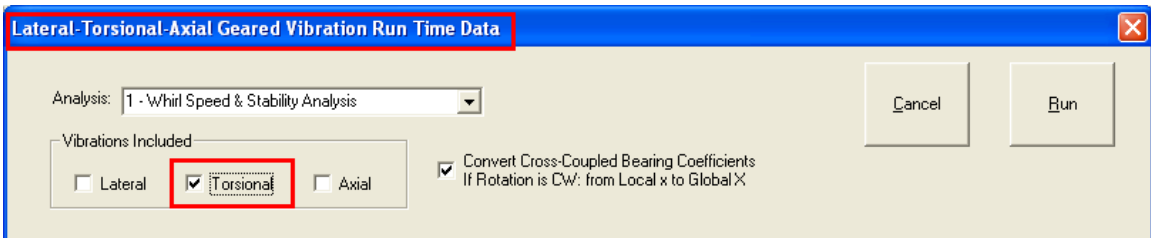
Manual Scaling Data:

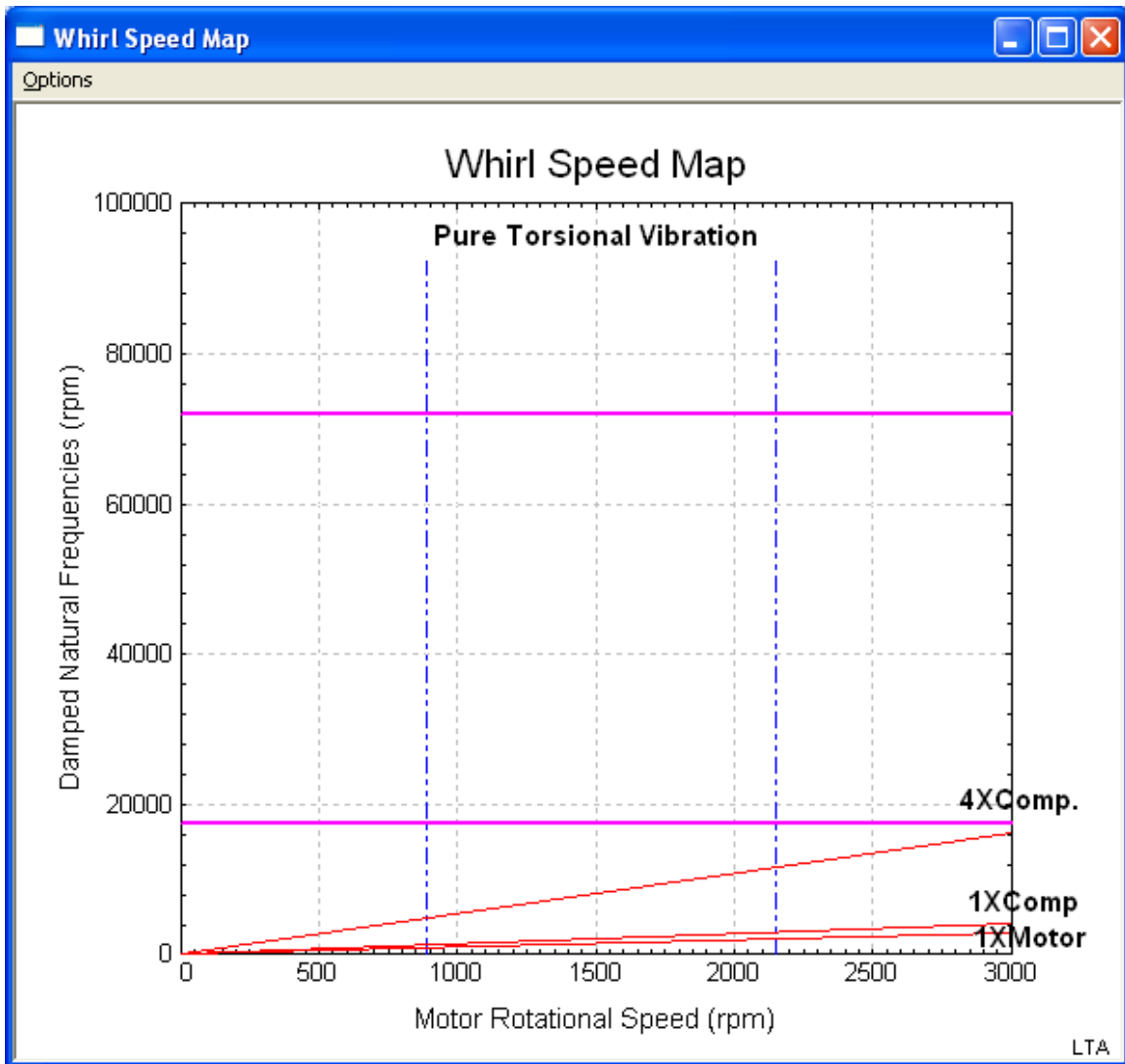
Xmin: 0	Ymin: 0	X Scale: 1.347
Xmax: 3500	Ymax: 30000	Y Scale: 1
XDiv: 7	YDiv: 6	



From the mode shape shown above, it indicates that this mode is a coupled lateral and torsional mode with very large torsional motion at the compressor and significant lateral motion at the motor driver end. Consequently, torsional excitation from the compressor rotor can cause high lateral vibration in the motor gear end, mainly in the Y direction (line of action) and small vibration in the X direction. With the natural frequency of 9660 cpm, the strong 4X response occurs at the compressor rotor speed of $9660/4=2,415$ rpm and motor speed of $9660/5.388=1793$ rpm.

To further study this system, in the Lateral-Torsional-Axial vibration run time input, we can check the Torsional vibration only. This will only analyze the torsional vibration with the flexibility in the gear meshes.





In summary, only the coupled lateral-torsional vibration analysis produces the interference point (potential resonance point) in the whirl speed map (Campbell Diagram). All the individual analysis on the lateral vibration and purely torsional analysis do not have interference point. Since the gyroscopic effect is very small and the frequencies stay almost the same as the speed increases, the following table summarizes the first several frequencies at zero rotor speed for various analyses:

Frequency (cpm) Comparison @ zero rotational speed						
Mode	Torsional (Rigid Link)	Torsional (Flexible Link)	Lateral Motion (Motor)	Lateral Motion (Male Rotor)	Lateral Motion (Female Rotor)	Coupled Vibration
1			12,905 (X)			9,660 (T,Y)
2			13,933 (Y)			12,978 (X,T)
3	18,857 (T)	17,549 (T)				14,537 (Y,T,X)
4				26,749 (X)		26,749 (X)
5			27,794 (X)			27,816 (X)
6			27,942 (Y)			33,580 (X)
7					33,580 (X)	34,255 (T,Y)
8			35,134 (X)			35,128 (X)
9			38,438 (Y)			38,347 (Y)
10				47,137 (X)		39,201 (Y,X,T)
11				47,247 (Y)		48,171 (X,Y,T)
12					49,434 (Y)	48,278 (Y,T)
13	72,146 (T)	72,031 (T)			62,232 (X)	62,232 (X)
14			73,776 (X)			69,440 (X,Y,T)
15			76,421 (Y)			73,775 (X,T)

Note: T indicates Torsional motion, X lateral motion mainly in X direction, Y lateral motion mainly in Y direction.

In view of the above table and the corresponding mode shapes, the frequency of the motor lateral motion mode (13,933 cpm) in the Y direction (line of action) decreases significantly due to the flexibility introduced by the gear mesh to become the lowest natural frequency (9,660 cpm) of the complete gear train. The frequency in the X direction (12,905 cpm) of the motor lateral motion is nearly unaffected by this torsional coupling effect and becomes 12,978 cpm in the coupled system.

To further study the torsional excitation effect on the rotor lateral vibration, the torsional excitation caused by the compressor is added into the model. The compressor has a 4/6 lobes design for the male and female rotors. It is known that it produces a strong 4X (male rotor speed) port passing torsional excitation from the compressor. This excitation is entered from Model – Data Editor – Torsional/Axial – Steady State Excitation as shown below. Note that the excitations are applied at shaft 2 (compressor male rotor), the excitation frequency is 4X of the compressor male rotor speed, not the motor speed. Also, the exciting torque increases with the square of the compressor speed.

Rotor Bearing System Data

Units/Description | Material | Shaft Elements | Disks | Unbalance | Bearings | Supports | Foundation | User's Elements
 Axial Forces | Static Loads | Constraints | Misalignments | Shaft Bow | Time Forcing | Harmonics | Torsional/Axial

Torsional and/or Axial Data

Linear Connectivity

Non-Linear Couplings/Connections

Modal Damping

Steady State Excitation (Single Harmonic)

Torsional Time Dependent Excitations

Excitations in Equations | I.C.

Excitations from Files

Torsional Startup Transient Torques

Motor Driving Torque

Load Torque

Reciprocating Excitation (nX Harmonics)

Engine Excitation

Recip. Torque

Lateral-Torsional-Axial Geared Coupling

Gear Mesh Data

Save | Save As | Close | Help

Torsional/Axial Steady State Excitation

Excitation Freq.(cpm): w0: 0 w1: 4 w2: 0

Torque/Force Multiplier: A0: 0 A1: 0 A2: 0.00016

$cpm = w0 + w1 \times rpm + w2 \times rpm^2$ $A = A0 + A1 \times rpm + A2 \times rpm^2$

Steady State Harmonic Excitation: $T = A * (Tc \cos(wexcT) + Ts \sin(wexcT))$

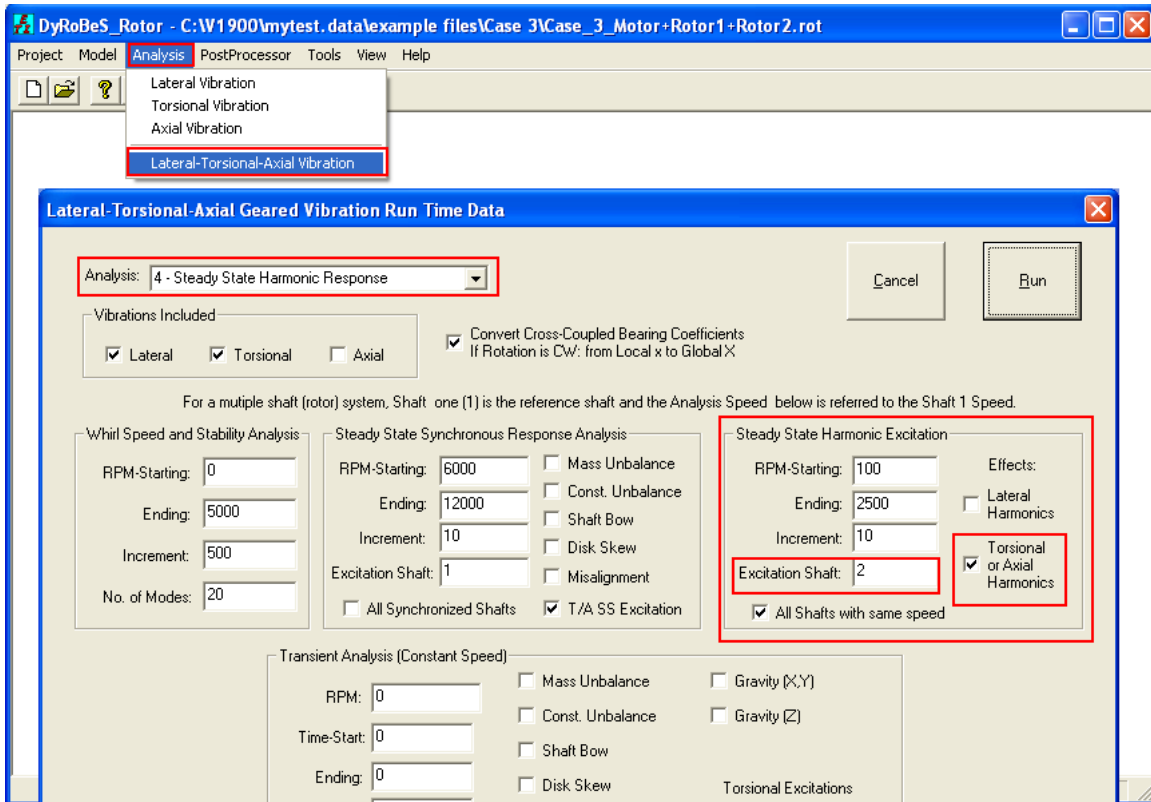
Excitation frequency $wexc$ (rad/sec) = $cpm * (2\pi/60)$ and A is the Amplitude multiplier

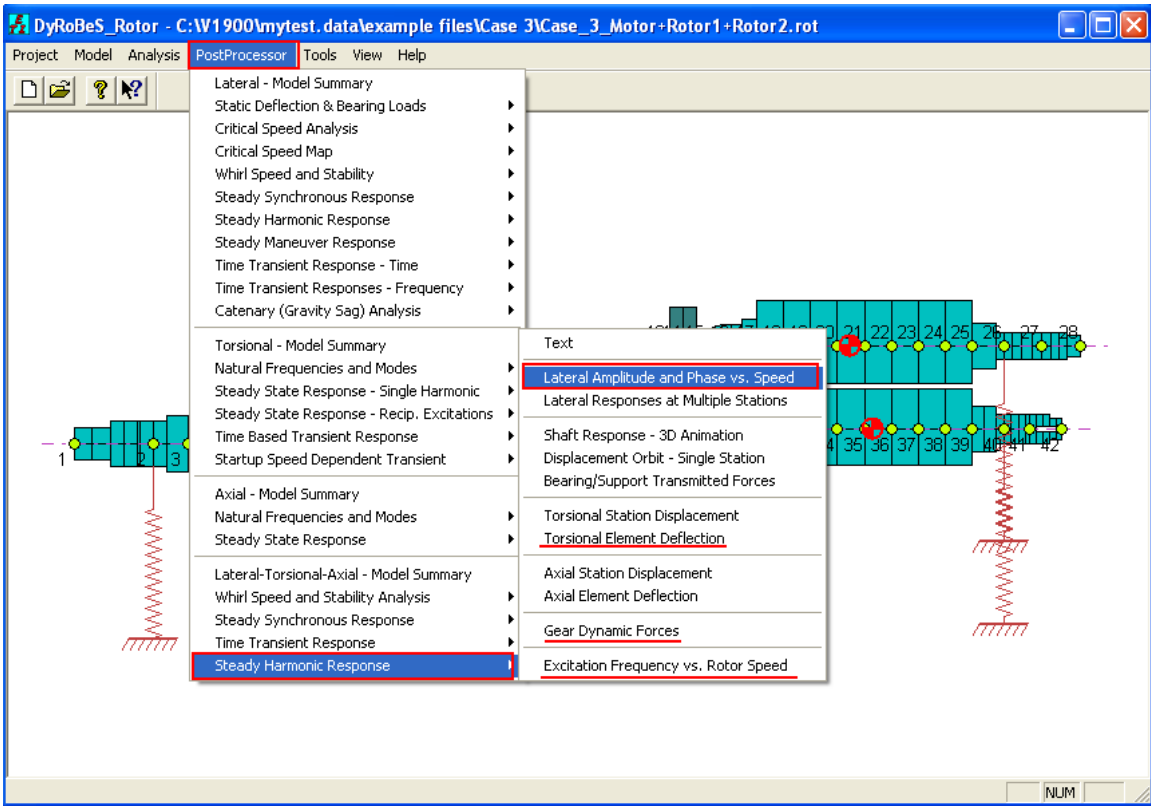
rpm = excitation shaft speed, rotor speed where the excitation applied

	T/A	Stn	Cos Component	Sin Component	Comments
1	Torsional	19	1	0	4 X Male Rotor Speed
2	Torsional	20	1	0	
3	Torsional	21	1	0	
4	Torsional	22	1	0	
5	Torsional	23	1	0	
6					
7					
8					
9					
10					
11					
12					

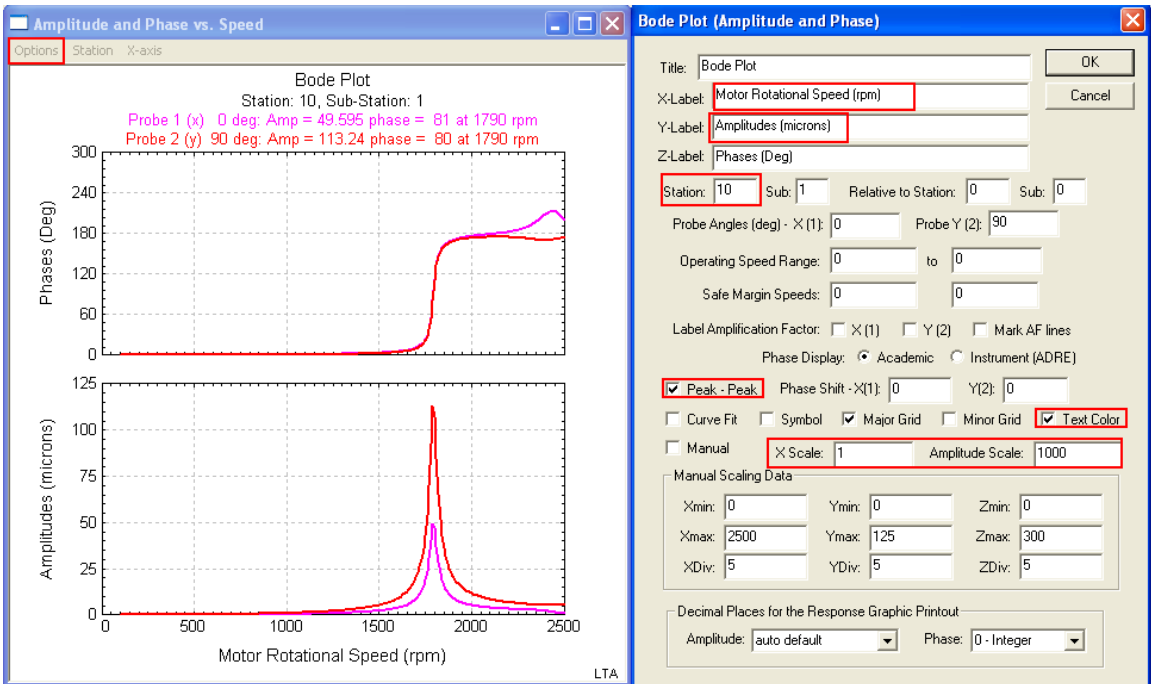
Unit(4) - T: Torque: N-mm; A: Force: N

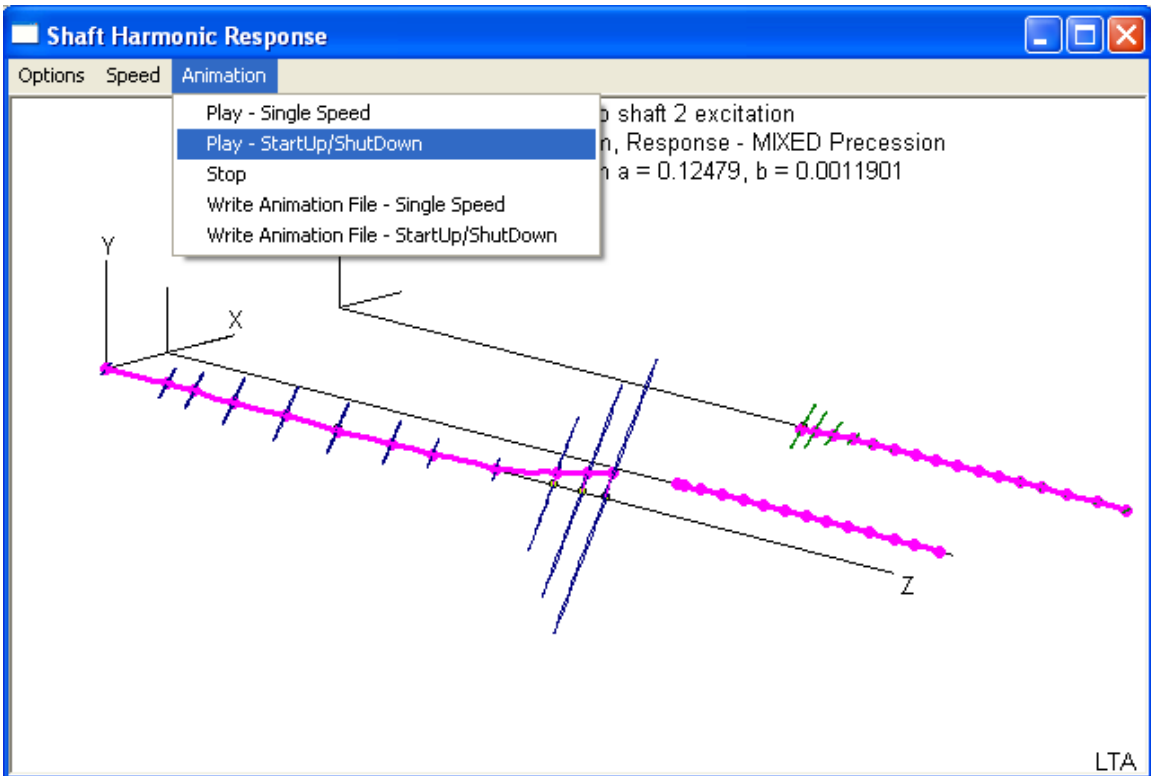
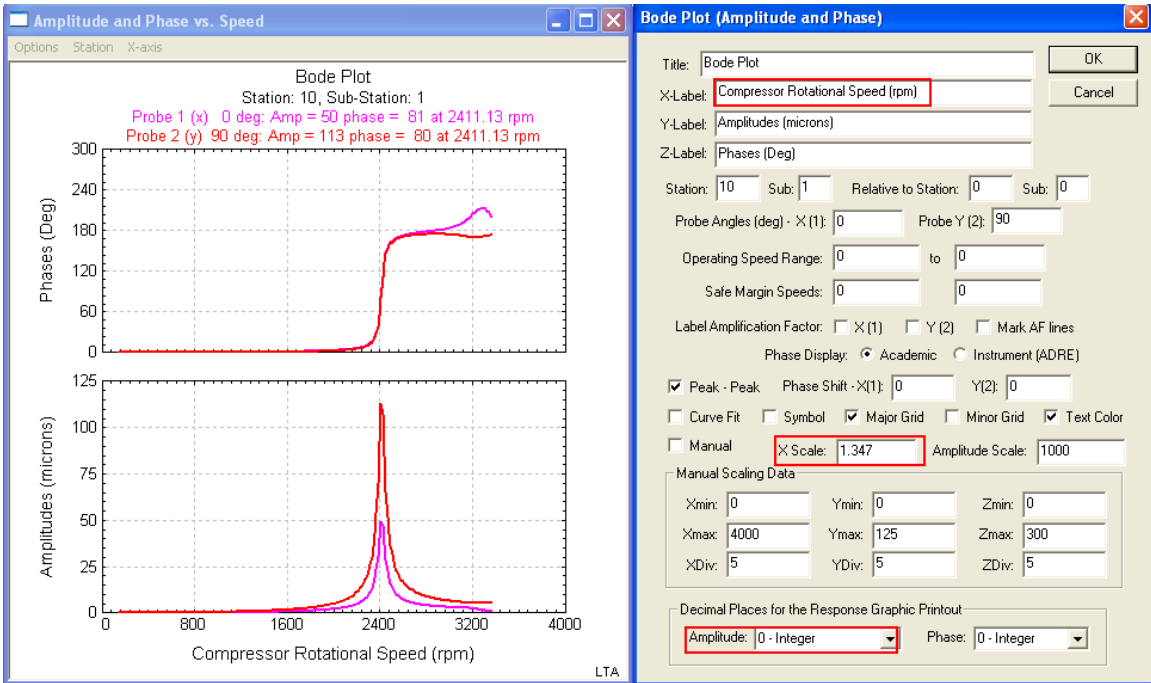
When performing the coupled lateral-torsional analysis, select Lateral-Torsional-Axial Vibration from the Analysis menu and enter the data below. Note that the analysis rotor speed is always referred to the shaft 1 rotor speed. Since the excitation is on shaft 2 (compressor male rotor), enter 2 in the excitation shaft field. Check the Torsional Harmonics to include the torsional harmonic excitation in the analysis. Several postprocessor plots are shown below. Although the excitation is due to torsional torque, the lateral vibration can be significant in this coupled gear system.

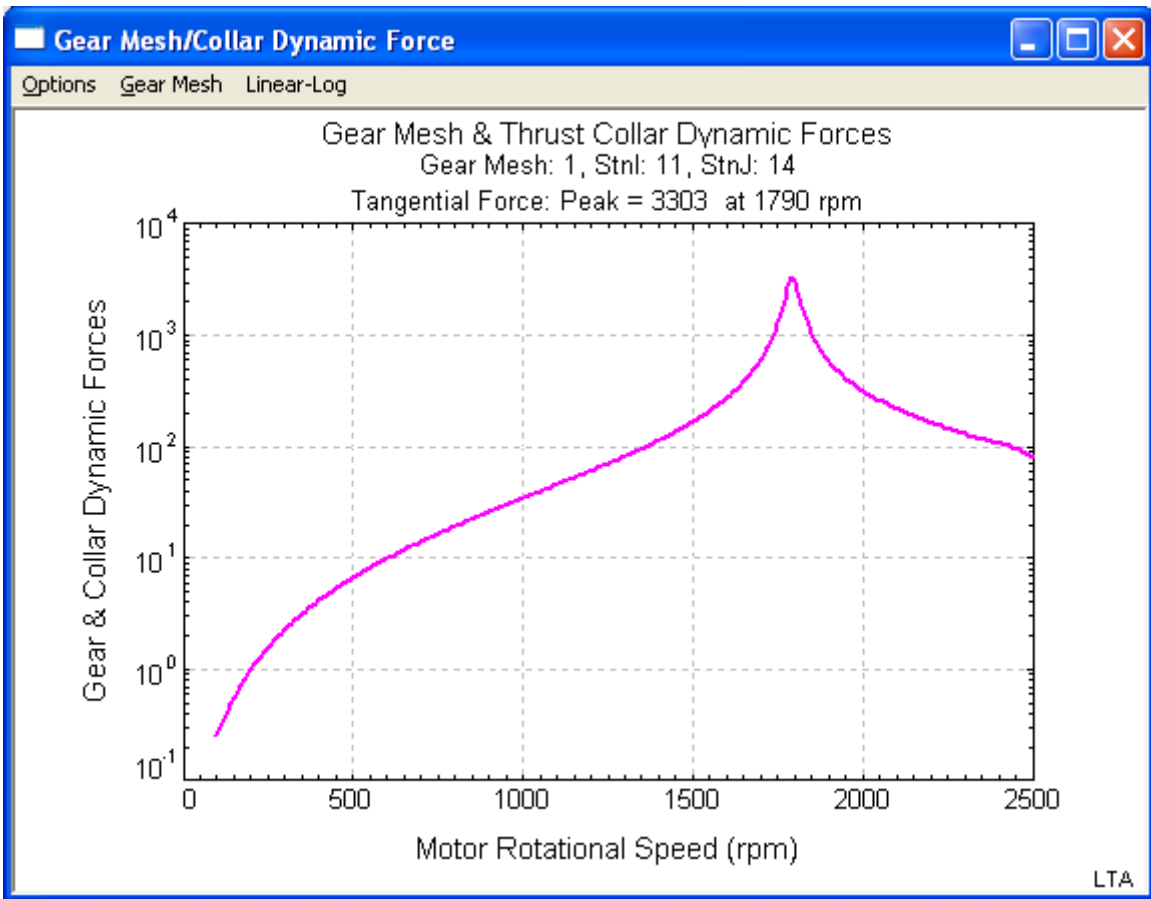




Again, by default, the rotor speed shown in the plot is the analysis speed (shaft 1 speed). To plot the vibration vs. the compressor male rotor speed, we can multiply the X-axis by 1.347.





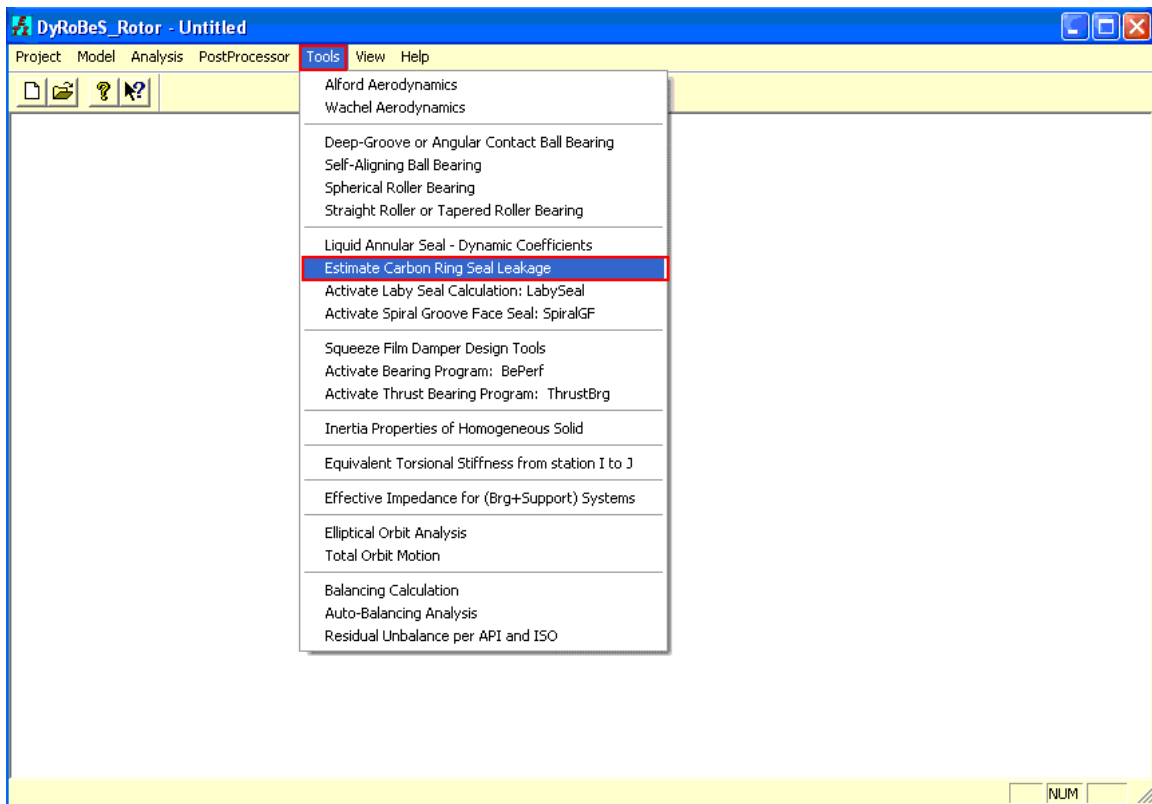


Carbon Ring Seal Calculation

The feature calculate the leakage for a carbon ring seal commonly used in the compressors and blowers. The calculation method is based on the air force report by Bauer et al. (1965). The original equations were adjusted with some correction factors to fit the experimental data tested by AMS Seals Inc.

Bauer, P., Glickman, M., and Iwatsuki, F., May 1965, "Analytical techniques for the Design of Seals for Use in Rocket Propulsion Systems, Volume II Dynamic Seals," Technical Report AFRPL-TR-65-61, Air Force Rocket Propulsion Laboratory, Research and Technology Division.

To activate this feature, Rotor – Tools – Estimate the Carbon Ring Seal Leakage.



Carbon Ring Seal Leakage Calculation X

Comment:

Units:

Total Seal Width: in

Shaft Diameter: in

Seal Diameter: in

High Pressure: psia

Low Pressure: psia

Speed (rpm):

Gas Properties

Gas Type:

Temperature: degF

Viscosity: Lbf-s/in²

Gas Const.: ft-lbf/lbm-R

Eccentricity Ratio:

Results:

Radial Clearance (C): 2C/D X 1000:

Laminar Flow Model

Vol. Flow: SCFM

Mass Flow: Lbm/min

Turbulent Flow Model

Vol. Flow: SCFM

Mass Flow: Lbm/min

The Maximum Allowable Residual Unbalance per API and ISO Specifications

The feature calculates the amount of maximum allowable residual unbalance in the rotor per API and ISO specifications. For more details, see book “Practical Rotordynamics and Fluid Film Bearing Design”, page 400-401.

To activate the function, Rotor –Tools – Residual Unbalance per API and ISO.

